

# LS-DYNA ${ }^{\circledR}$ Database Binary Output Files 

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## INTRODUCTION

Three databases are discussed, these are:

1. State Database ( $G=p t f$, default name d3plot and d3part)
2. Time History Database ( $\mathrm{F}=\mathrm{thf}$, default name d3thdt)
3. Interface Force Database ( $\mathrm{S}=\mathrm{iff}$, no default name, typically: intfor)

The purpose of this information is to give guidance on how to access and read the various databases.
The databases are written as word addressable fixed length binary files. The actual length depends on the amount of data saved, but will always be a multiple of 512 words ( 4 or 8 bytes each). Since it is likely that the database cannot be contained in a single file of length, FEMLEN, the data will spread over several files known as a family of files. Having a set of files enables them to be handled more easily than a single very large file. The root name for a family is the name of the first file member. Successive member names are compiled by appending a two or three digit number to the root name starting with 01,02 , through to 99 , then 100 and ending with 999. Root names are limited to 75 characters. The original reason for a family of files was because the hard disks used for dyna3d runs could not cope with a single contiguous files large enough to contain all the data output. Subsequently, it has been found that splitting the output into separate files allows some unwanted data to be discarded and aids the copying, saving and movement of the data. Very large files can be impossible to transfer reliably over networks. Total output data can amount to several gigabytes or more depending on the model size.
For ls-dyna runs with mesh adaptivity, the root name has a two letter appendage for each adapted mesh. Starting from 'aa' through 'az', then 'ba' through 'bz' and continuing up to 'zz', this gives a maximum of 676 possible adaptions. For example if the root name is 'd3plot' the subsequent files related to the original mesh are 'd3plot01', d3plot02, ..., after adaption the new mesh and undeformed geometry is put in 'd3plotaa' and the subsequent files for the new mesh are: 'd3plotaa01', d3plotaa02, ... The next adapted mesh will be in file 'd3plotab' and so on. A set of files at a particular adaption can be read separately by giving the root name with appendage, as the base file name.
For example command: 'Isprepost d3plot' will read in all the file with root name 'd3plot' While: 'Isprepost d3plotab' will read in only files have 'd3plotab' in the name.
LSPREPOST will read the binary databases separately or combined.
Eg: 1sprepost d3plot, 1sprepost d3thdt, lsprepost iffname, 1sprepost d3plot $\mathrm{h}=\mathrm{d} 3$ thdt $\mathrm{f}=\mathrm{iffname}$

The file length used is set in the ls-dyna run as the default size of $7 \times 512 \times 512$ words. The size can be changed on the command line with the ' $\mathrm{x}=$ factor' parameter giving a size of: factorx $512 \times 512$ words.
If the initial data or state data is larger than the given file length, the data will automatically split across files. This condition is not desirable because it is not clear whether any non root file can be discarded without destroying the continuity of the data. Ls-dyna checks before writing to a file, to ensure that there is room left in the file to contain the data at a particular state time. If not, it closes the current family member and starts writing the state data in the next file member. The files are written with a block size of 512 words, and if the data does not complete the last block it is padded out. This means that files cannot be concatenated and read together. The word size is 4 bytes for the single precision version of ls-dyna and 8 bytes for the double precision version, unless 32bit ieee format is defined, see *DATABASE_FORMAT, IBINARY.

## FILE GENERAL STRUCTURE

The root file starts with a control words section, followed by node coordinates, then element connectivity for solids, thick shells, beams, and shells. Next are lists to reference the sequential internal numbering to the users number. State data is output next always starting with the time word. Data is of fixed length through the file members except where the mesh is adapted. The length of each area can be calculated from the information in the control words. The first file at adaption is like the root file in structure, so the new control words are used to recalculate the size of the subsequent data. The root file contains the initial data and also state data if there is room to write it. Further state data is written to the family members and each file will start with the time word provided data from the previous state did not overflow onto the file.

If the disk address, DA, of the data being written exceeds the maximum file length, then data is written into file number int(DA/FAMLEN) at location DA-FAMLEN*int(DA/FAMLEN). If the state length is greater than the remaining length, the disk address is increased to start the writing at the beginning of the next file.

## STATE DATABASE (d3plot)

There are three sections in this database. The first contains 64 words of control information plus extensions. The second contains geometric information including the nodal coordinates and element connectivities and user numbering lists. The third section contains the results of the analysis at sequential output intervals. The output at a given time, called a state, contains a time word, global variables such as total energies and momenta for the whole model and each material (part), node data consisting of displacements, velocities, accelerations, and optionally temperatures, and finally element data that can include stresses and strains at integration points, and element deletion flags. The control data provides information about what is in the file and is used to calculate the various data length.
There are two other state database files, namely: d3drfl and d3part, these are similar to d3plot but contain less data. The dynamic relaxation file, d3drfl, provides the state at the end of the DR process, while d3part is state output for a reduced number of parts in the model.

## CONTROL DATA



| Version | 1 | 14 | Code version, floating number, eg 960.0 it is used to distinguish the floating point format, like cray, ieee, and dpieee |
| :---: | :---: | :---: | :---: |
| NDIM | 1 | 15 | Number of dimensions ( 2 or 3 ). If 5 or 7 then an array of material types is read (MATTYP $=1$ ), element connectivities are unpacked and NDIM=3. If 4 then element connectivies are unpacked in the DYNA3D database and NDIM is reset to 3 . If $>5$ and $<8$ then state data contains movement of rigid road surface. If NDIM $=8$ or 9 , coordinates, velocities and accelerations are not output for rigid bodies nodes. At the end of each state motion data is output for each rigid body. See below for more details. If $=9$ rigid road surface data is included. |
| NUMNP | 1 | 16 | Number of nodal points |
| ICODE | 1 | 17 | Flag to identify finite element code=2: old DYNA3D, code=6: NIKE3D, LS-DYNA/3D, LS-NIKE3D database |
| NGLBV | 1 | 18 | Number of global variable to be read with each state NUMRW=number of rigid walls. <br> NUMRBS=number of rigid body sets. <br> $=6+7$ * (NUMMAT8 + NUMMAT2 + NUMMAT4 + <br> NUMATT+NUMRBS) + NUMRW * N <br> $\mathrm{N}=1$ for DYNA3D and LS-DYNA3D <br> $\mathrm{N}=4$ for LS-DYNA >= version 971 |
| IT | 1 | 19 | Flag for temperatures <br> $=0$, none, <br> $=1$, read in a temperature for each node <br> $=2$, read temperature for each node and heat flux for each node. <br> $=3$, read thermal shell middle temperature, thermal shell inner temperature, thermal shell outer temperature, and heat flux for each node. Solid node temperatures are repeated <br> $+=10$, read mass scaling value for each node |
| IU | 1 | 20 | Flag for current geometry ( $=1$ or 0 ) |
| IV | 1 | 21 | Flag for velocities ( $=1$ or 0 ). If $<0$ d3eigv file contains unscaled data. |
| IA | 1 | 22 | Flag for accelerations ( $=1$ or 0 ) |

NEL8

NUMMAT8
NUMDS
NUMST
NV3D

NEL2
NUMMAT2
NV1D
NEL4
NUMMAT4
NV2D

NEIPH

NEIPS

MAXINT

1

1

1

1

1
1

1

1
1

1

1

1
35

1
36

Number of 8 node solid elements
If NEL8 $<0,2$ extra nodes are output for ten node solids. Array is 2 * abs(NEL8), and is read after the arbitrary numbering arrays.

Number of materials (parts) used by the 8 node solids

If $<0$, Shell element data is output for 4 in-plane Gauss points. MAXINT will be 4 times number of points through the plane.
$=0$ for d3plot/d3part
Number of values in database for each solid element. $=6 *$ IOSOL(1) + IOSOL $(2)+$ NEIPH
If NV3D $>=8$ * ( 6 *IOSOL(1)+IOSOL(2)+NEIPH), each solid element has values at each Gauss point.
Number of 2 node one-dimensional elements
Number of materials (parts) used by the 2 node 1D elements

Number of values in database for each 1D element $=6+5 *$ BEAMIP + NEIPB $*$ ( $3+$ BEAMIP ).

Number of four node shells (2D or 3D) elements.
Number of materials (parts) used by the 4 node 2D elements
Number of values in database for each 2D element Are: $\mathrm{MAXINT}^{*}(6 * \operatorname{IOSHL}(1)+\mathrm{IOSHL}(2)+\mathrm{NEIPS})+8$ *IOSHL(3)+4*IOSHL(4)+12*ISTRN

Number of additional values per integration point in a solid element, see NV3D. NEIPH is the number of values for history variables + total strains + plastic strains + thermal strains. ISTRN and IDTDT specifies which strains are present.

Number of additional values per integration point to be written into the type 6 database for shell elements.

Number of integration points dumped for each shell and the MDLOPT flag. The magnitude of MAXINT will be greater than or equal to 3 .
if MAXINT $>=0$, then MDLOPT $=0$ and MAXINT=MAXINT
if MAXINT $<0$ then MDLOPT $=1$ and MAXINT $=$ abs(MAXINT)
if MAXINT $<-10,000$, then MDLOPT $=2$ and

MAXINT=abs(MAXINT)-10,000
MDLOPT controls the element deletion table (see below). This data allows deletion by nodes or elements.

| EDLOPT | 1 | 37 | Element deletion flag (not standard) <br> $=x x x 1$ Solids deleted <br> $=x x 1 \times$ Beams deleted <br> $=x 1 x x$ Shells deleted <br> $=1 \mathrm{xxx}$ Thick Shells deleted <br> (Not used in LS-DYNA) |
| :---: | :---: | :---: | :---: |
| NMSPH | 1 | 37 | Number of SPH Nodes |
| NGPSPH | 1 | 38 | Number of SPH materials |
| NARBS | 1 | 39 | Additional storage required for arbitrary node and element numbering in type 6 database $=0$, Sequential numbering. |
| NELT | 1 | 40 | Number of 8 node thick shell elements. MAXINT*(6*IOSHL(1)+IOSHL(2)+NEIPS) + $12 *$ ISTRN |
| NUMMATT | 1 | 41 | Number of materials (parts) used for the 8 node thick shell element. |
| NV3DT | 1 | 42 | Number of values in database for each thick shell |
| IOSHL(1) | 1 | 43 | $\begin{aligned} & 6 \text { stress components flag. } \\ & \text { if 1000: } \\ & \text { if } 999: \quad \operatorname{IOSHL}(1)=1, \operatorname{IOSOL}(1)=1 \\ & \text { else: } \\ & \text { IOSHL }(1)=0, \operatorname{IOSOL}(1)=1 \\ & \text { eloSOL }(1)=0 \end{aligned}$ |
| IOSHL(2) | 1 | 44 | Plastic strain flag. <br> if 1000: $\operatorname{IOSHL}(2)=1, \operatorname{IOSOL}(2)=1$ <br> if 999: $\operatorname{IOSHL}(2)=0, \operatorname{IOSOL}(2)=1$ <br> else: $\operatorname{IOSHL}(2)=0, \operatorname{IOSOL}(2)=0$ |
| IOSHL(3) | 1 | 45 | Shell force resultants flag, if $1000=1$ else $=0$ |
| IOSHL(4) | 1 | 46 | Shell thickness, energy +2 others, if $1000=1$ else $=0$ |
| IALEMAT | 1 | 47 | Size of array containing solid element parts numbers used as ALE material |
| NCFDV1 | 1 | 48 | Bit flags for CFD nodal values, or if $=67108864$, then state contains Multi-Solver extra data - see formats descriptions below. |


| NCFDV2 | 1 | 49 | Further bit flags for CFD nodal values or if MS extra data, then value equals number of data domains. |
| :---: | :---: | :---: | :---: |
| NADAPT | 1 | 50 | Number of adapted element to parent pairs (not implemented) |
| NMMAT | 1 | 51 | Total number of materials (parts) - not set in LSDYNA/3D |
| NUMFLUID | 1 | 52 | Total number of ALE fluid groups. Fluid density and volume fractions output as history variables, and a flag for the dominant group. If negative multi-material species mass for each group is also output. Order is: rho, vf1, ... vfn, dvf flag, $\mathrm{ml}, \ldots \mathrm{mn}$. Density is at position 8 after the location for plastic strain. Any element material history variables are written before the Ale variables, and the six element strains components after these if ISTRN=1. |
| INN | 1 | 53 | Invariant node numbering fore shell and solid elements See INN in card *CONTROL_ACCURACY |
| NPEFG | 1 | 54 | Number of particle method data sets. When the seventh digit is set to 1 there is Discrete Element Sphere (DES) output, i.e. xx 1 xxxxx. NPEFG/ $1000000=1$. |
| NEL48 | 1 | 55 | Number of 8 node Shells. Internal element id and 4 extra nodes are output for each 8 node shell. Array is 5 * NEL48 long, and is read before the Header, Part and Contact details, and after the extra nodes for 10 nodes solids if they exist. |
| IDTDT | 1 | 56 | Flags for various data in the database, these are: unit, tenth digit, hundredth, and so on. Extract the digit from IDTDT and interpret as following types: |
|  |  |  | IDTDT $/ 1=1$ : <br> An array of dT/dt values of length NUMNP. Array is written after node temperature arrays. (xxxx1) |
|  |  |  | IDTDT $!10=1$ : <br> An array of residual forces of length $3 *$ NUMNP followed by residual moments of length $3 *$ NUMNP. This data is written after node temperatures or $\mathrm{dT} / \mathrm{dt}$ values if there are output. (xxx1x) |

$\left.\begin{array}{ll}\text { IDTDT }!100=1: & \begin{array}{l}\text { Plastic strain tensor is written } \\ \text { for each solid and shell after } \\ \text { standard element data. For } \\ \text { solids }(6 \text { values) and for shells } \\ (6 \times 3=18 \text { values), at the }\end{array} \\ \text { lower, middle and upper } \\ \text { integration location. (xx1xx) }\end{array}\right\}$ IDTDT ! 1000=1: $\left.\quad \begin{array}{l}\text { Thermal strain tensor is written } \\ \text { after standard element data. For } \\ \text { solid (6 values) and shell (6 } \\ \text { values) and after the plastic } \\ \text { strain tensor if output. (x1xxx) }\end{array}\right\}$

| EXTRA 1 | 57 | Additional number of control words. If $>0$ 0, there are <br> EXTRA control words after the first 64 words. |
| :--- | :--- | :--- | :--- |

WORDS 6 58-63 Used by D3THDT and INTFOR

The value of ISTRN must be computed if IDTDT $<100$, because then it is not output in the control data. For this case only, this is the rule for computing it.

ISTRN can only be computed as follows and if NV2D $>0$.
If NV2D-MAXINT*(6*IOSHL(1)+IOSHL(2)+NEIPS)+8*IOSHL(3)+4*IOSHL(4) > 1
Then ISTRN $=1$, else ISTRN $=0$
If ISTRN $=1$, and NEIPH $>=6$, last the 6 additional values are the six strain components.
Or NELT > 0
If NV3DT-MAXINT*(6*IOSHL(1)+IOSHL(2)+NEIPS) $>1$
Then ISTRN $=1$, else ISTRN $=0$

## If (EXTRA>0) Extra control words are written after the first 64 words

(This is currently set to 64 and 2 are used from March 2014, remaining are set to zero)
DISK
VALUE \#WORDS ADDRESS DESCRIPTION
NEL20 1664 Number of 20 Node Solid Elements. Internal element id and 12 extra nodes are output for 20 node solids. Array is 13 * NEL20 long, and is read before Header, Part and Contact details, and after the extra nodes for 8 node shells, if they exist.


## MATERIAL TYPE DATA

The material section contains the material type numbers. This section is skipped if MATTYP is zero.
This data is required because those shell elements that are in a rigid body have no element data output in the state data section. The normal length of the shell element state data is:
NEL4 * NV2D, when the MATTYP flag is set the length is: (NEL4 - NUMRBE) * NV2D. When reading the shell element data, the material number must be check against IRBRTYP list to find the element's material type. If the type $=20$, then all the values for the element to zero. This option is set in *DATABASE_EXTENT_BINARY, with DCOMP=2

## VALUE

NUMRBE

NUMMAT
IRBTYP

LENGTH

1
1
NUMMAT

## DESCRIPTION

Number of rigid body shell elements.
Number of materials in the database.
Material type numbers

## FLUID MATERIAL ID DATA

The fluid material section contains the material numbers for solid elements that are used to define an Euler grid or Arbitrary Lagrangian Euler mesh. This section is skipped if IALEMAT is zero.

VALUE
FLUIDID

## LENGTH <br> DESCRIPTION

IALEMAT

Fluid material number used in solid element mesh

## SMOOTH PARTICLE HYDRODYNAMICS ELEMENT DATA FLAGS

This section is only output if NMSPH $>0$. The section is a list of flags to indicate what SPH data is output for each SPH node/element. The first number is the length in words for this array, currently $=11$.
SPH elements are centered at nodes, and cover a spherical volume defined by the radius of influence. They do not have a connection with other SPH elements. They should be displayed as a dot or a spherical surface, with radius scaling to reduce the size and enable each element to be distinguishable.

As follows:

$$
\begin{aligned}
& \text { isphfg }(1)=11 \text { - length of sph flags array } \\
& \operatorname{isphfg}(2)=1 \quad \text { - radius of influence } \\
& \operatorname{isphfg}(3)=1 \quad \text { - pressure in particle } \\
& \operatorname{isphfg}(4)=6-6 \text { true stress components } \\
& \text { isphfg(5) }=1 \quad \text { - plastic strain, }>0.0 \text { if effective stress exceeds yield strength } \\
& \operatorname{isphfg}(6)=1 \quad \text { - density of particle material } \\
& \operatorname{isphfg}(7)=1 \quad \text { - internal energy (strain) } \\
& \text { isphfg }(8)=1 \quad \text { - number of neighbors affecting particle } \\
& \text { isphfg }(9)=6-6 \text { true strain components } \\
& \operatorname{isphfg}(10)=1 \quad-\text { mass of element }(>=1 \mathrm{~s} 971) \\
& \operatorname{isphfg}(11)=1 \quad-\max \text { number of sph history variables. }
\end{aligned}
$$

If any value of $\operatorname{isphfg}(2)$ through $\operatorname{isphfg}(11)=0$, then the particular data item is not output for the particle. To calculated the size of data add the isphfg values from isphfg(2) through
isphfg(11) plus one. One value is always output which is the material number as a floating point number for each particle.

## If this value is negative then the particle has been deleted from the model.

Note: it is possible a SPH element could be deleted, or be non active in the initial states, and become active in later states.
Full output for each particle is:
mat\#, radius, pressure, \{sx, sy, sz, sxy, syz, sxz\} ps, rho, ie, nn, \{ex, ey, ez, exy, eyz, exz\}, mass, hv1 ... hvn.
NUM_SPH_VARS = $1+\operatorname{sum}$ of $\operatorname{isphfg}(\mathrm{i}), \mathrm{i}=2$ to isphfg(1)
Hence, total size is $20+$ the total number of history variables.
When a particle is deleted from the model, data is still output for it because the length of data must always be the same for each state.

## PARTICLE DATA (NPEFG $>0$ )

DES Control Block - see description below

## Control Block

If NPEFG $>0$ and NPEFG $<10000000$, airbag particles are output
The first three digits of NPEFG are the number of airbags in the database $=$ NPARTGAS
NPARTGAS $=$ NPRFG $\% 1000$
SUBVER = NPEFG / 1000
In the extended control block:
The first four words in the block are:

1. NGEOM number of geometry variables
2. NVAR number of state variables
3. NPART number of particles
4. NSTGEOM number of state geometry variables

If SUBVER $==4$
5. NCHAMBER number of chambers

NLIST $=$ NGEOM + NVAR + NSTGEOM

NLIST words of output for variables listed to define the type of each variable, $=1$ for integer and $2=$ for floating point
$8 *$ NLIST words of 8 character variable names (each integer word is an ascii character).

## GEOMETRY DATA

The geometry section contains the nodal coordinates and the element connectivities. The ordering of the nodal points is the same as the ordering of the nodal data in the state data that follows. If NDIM=3 the connectivities are assumed to be packed with 3 integers per word, if NDIM $>3$, then connectivities are not pack, (the default for LS-DYNA, LS-DYNA3D and LSNIKE3D. The order of the elements are 3, 2, and 1 dimensional elements if the database is ICODE=2 or 6 .

| VALUE | LENGTH | DESCRIPTION |
| :---: | :---: | :---: |
| $\mathrm{X}(3,1)$ | NDIM*NUMNP | Array of nodal coordinates $\mathrm{X} 1, \mathrm{Y} 1, \mathrm{Z} 1$, X2,Y2,Z2, X3, Y3,Z3, ... ,Xn, Yn,Zn |
| IX8(9,1) | 9*NEL8 | Connectivity and material number for each 8 node solid element. |
| If NEL8 < 0 | 2*abs(NEL8) | Extra nodes for ten node solids - not read here. |
| IXT $(9,1)$ | 9*NELT | Connectivity and material number for each 8 node thick shell element. |
| IX2 $(6,1)$ | 6*NEL2 | Connectivity, orientation node, two null entries, and the material number for each 2 node beam element. <br> For some beam types the last two number contain the beam type and length to width ratio * 100 and length to height ratio * 100 type $=\mathrm{ix} 2\left(5,{ }^{*}\right) \& 0 \times 3 \mathrm{~F}$ width $=0.01$ * length $/\left(\operatorname{ix} 2\left(5,{ }^{*}\right) \gg 6\right.$ height $=0.01 *$ length $/ \mathrm{ix} 2\left(6,{ }^{*}\right)$ Third node (orientation) may be $>1 \mathrm{e} 9$ Contain flag 1e9 to indicate a spot weld. |
| IX4(5,1) | 5*NEL4 | Connectivity and material number for each 4 node shell element |

Note the node numbers are the LS-DYNA internal numbers for nodes, these will be the same as the user's numbers if NARBS $=0$, otherwise, the arbitrary number lists are used to find the user's numbers, similarly, for element numbers and material numbers.

## USER MATERIAL, NODE, AND ELEMENT IDENTIFICATION NUMBERS

Skip this section if NARBS (disk address 39) is zero. The user node and element numbers must be in ascending order. It assumed that if this option is used all node and element data anywhere in the databases is in ascending order in relation to the user numbering. Read in NARBS words and decipher as indicated below.

For sequential material/part numbering, the total length of data is:
NARBS $=10+$ NUMNP+NEL8+NEL2+NEL4+NELT +
3*NMMAT : these numbers are not used
Only the first 10 control words are read in and used. The other 6 words are only output by ls-dyna when NSORT $<0$.

For arbitrary material numbering (NSORT $<0$ )
NARBS $=16+$ NUMNP+NEL8+NEL2+NEL4+NELT+3*NMMAT
Where material numbers are not in ascending order.
For this case all 16 control words are read in.

## VALUE <br> NSORT

NSRH

NSRB

NSRS

NSRT

NSORTD 1
NSRHD
NSRBD 1
NSRSD
NSRTD

1
LENGTH
1

1

1

1

1


1

1

## DESCRIPTION

Pointer to arbitrary node numbers in LS-DYNA code. If $<0$, it flags that arbitrary material identification numbers are also used.
Pointer to arbitrary solid element numbers in LS-DYNA code:
$=$ NSORT + NUMNP
Pointer to arbitrary beam element numbers in LS-DYNA code: =NSRH+NEL8
Pointer to arbitrary shell element numbers in LS-DYNA code:
=NSRB+NEL2
Pointer to arbitrary thick shell element numbers in
LS-DYNA code:
=NSRS+NEL4
Number of nodal points
Number of 8 node solid elements
Number of 2 node beam elements
Number of 4 node shell elements
Number of 8 node thick shell elements
\(\left.$$
\begin{array}{lll}\text { NSRMA } & 1 & \begin{array}{l}\text { Pointer to an array in the LS-DYNA code that list } \\
\text { the material ID's in ascending order. }\end{array} \\
\text { NSRMU } & 1 & \begin{array}{l}\text { Pointer to an array in the LS-DYNA code that gives } \\
\text { the material ID's in the actual order that they are } \\
\text { defined in the user input. }\end{array} \\
\text { NSRMP } & 1 & \begin{array}{l}\text { Pointer to an array in the LS-DYNA code that gives } \\
\text { the location of a member in the array originating at }\end{array}
$$ <br>

NSRMU for each member in the array starting at\end{array}\right\}\)| NSRMA. |
| :--- |

## RIGID BODY DESCRIPTION for NDIM=8,9

| NRIGID | 1 | Number of rigid bodies |
| :--- | :--- | :--- |
| For each rigid body: |  |  |
| MRIGID | 1 | Rigid body part internal number |
| NUMNODR | 1 | Number of nodes in rigid body |
| NODER | NUMNODR | Internal node number of rigid body |
|  |  |  |
| NUMNODA | 1 | Number of active (not rigid) nodes |
| NODEA | NUMNODA | Internal node numbers of active nodes. |

Active node coordinates, velocities and accelerations are output as usual. Rigid body motion data is described in the state section below.

## ADAPTED ELEMENT PARENT LIST (not implemented)

List of element id pairs for H-type shell element adaptivity.
Length of data is $2 *$ NADAPT, pairs are element number and element parent number

## SMOOTH PARTICLE HYDRODYNAMICS NODE AND MATERIAL LIST

$\begin{array}{ll}\text { If NMSPH }>0 & \text { List of sph node and its material number } \\ \text { Length of data } & 2 * \text { NUMSPH }\end{array}$

## PARTICLE GEOMETRY DATA (NPEFG > 0)

DES Control words (NPEFG/10000000 ==1) - see description below

NPARTGAS blocks of NGEOM data to describe the geometry for each airbag:

1. first particle ID for the airbag
2. number of particles in the airbag
3. ID for the airbag
4. number of gas mixtures in the airbag

If $\mathrm{NGEOM}==5$
5. number of chambers

## RIGID ROAD SURFACE DATA

If NDIM > 5
NNODE Number of nodes in road surface
NSEG Total number of 4 noded road surface segments
NSURF Number of road surfaces
MOTION Flag to indicate motion data is output for each state
NODEID NNODE list of IDs
SURFNODE XYZ Coordinate for each node
Lists of 4 noded segments for each surface
SURFID Surface ID Number
SURFNSEG Number of segments in surface
SURFSEGS SURFNSEG of 4 node ids for each segment

Length of data $=4+$ NNODE $+3 *$ NNODE + NSURF $*(2+4 *$ SURFNSEG $)$

EXTRA 2 NODE CONNECTIVITY FOR 10 NODE TETRAHEDRON ELEMENTS (only if NEL8 < 0)List of extra nodes for each 10 node tetrahedron element, 2 * abs(NEL8). Any 8 node solids have these two nodes set to zero.

EXTRA 4 NODE CONNECTIVITY ARRAY FOR 8 NODE SHELL ELEMENTS
(only if NEL48 > 0)
NEL48 number of 8 node shells
List of extra nodes for each 8 node shell element, 5 * NEL48:- element internal number and 4 extra nodes.

EXTRA 12 NODE CONNECTIVITY ARRAY FOR 20 NODE SOLID ELEMENTS (only if EXTRA $>0$ and NEL20 $>0$ )
NEL20 number of 20 node solids
List of extra nodes for each 12 node solid elements, 13 * NEL20:- element internal number and 12 extra nodes.

## EXTRA 19 NODE CONNECTIVITY ARRAY FOR 27 NODE SOLID ELEMENTS (only if EXTRA > 0 and NEL27 > 0) <br> NEL27 number of 27 node solids

List of extra nodes for each 19 node solid elements, 20 * NEL27:- element internal number and 19 extra nodes.

EXTRA 13 NODE CONNECTIVITY ARRAY FOR 21 NODE SOLID ELEMENTS (only if EXTRA $>0$ and NEL21P >0)
NEL21P number of 21 node solids
List of extra nodes for each 13 node solid elements, 14 * NEL21P:- element internal number and 13 extra nodes.

## EXTRA 7 NODE CONNECTIVITY ARRAY FOR 15 NODE SOLID ELEMENTS <br> (only if EXTRA > 0 and NEL15T > 0) <br> NEL15T number of 15 node solids

List of extra nodes for each 15 node solid elements, 8 * NEL15T:- element internal number and 7 extra nodes.

EXTRA 12 NODE CONNECTIVITY ARRAY FOR 20 NODE SOLID ELEMENTS (only if EXTRA > 0 and NEL20T $>0$ )
NEL20T number of 20 node solids
List of extra nodes for each 12 node solid elements, 13 * NEL20T:- element internal number and 12 extra nodes.

## HEADER, PART \& CONTACT INTERFACE TITLES

At the end of the first binary files, eg d3plot, the part and model titles are appended.
If the model input includes *DATABASE_BINARY_D3PROP, all the d3prop part data is included.
At the end of the first interface force file, titles and contact id are appended.
This extra data is written at the end of the following files:
d3plot, d3part and intfor files, and the header and part titles are written directly after the EOF (= -999999.0) marker.

Header output

| ---------------------------------- |  |  |
| :--- | :--- | :--- |
| NTYPE | 1 | entity type $=90000$ |
| HEAD | 18 | Header title (72 characters) |

For the interface force file (intfor), header and contact titles are written at the end of first file after the EOF (= -999999.0) marker

Part title output

| Value | Length | Description |
| :--- | :--- | :--- |
| ----------------------- |  |  |
| NTYPE | 1 | entity type $=90001$ |
| NUMPROP | 1 | number of parts |

For NUMPROP parts:

| IDP | 1 | part id |
| :--- | :--- | :--- |
| PTITLE | 18 | Part title (72 characters) |

For the interface force file (intfor), header and contact titles are written at the end of first file after the EOF (=-999999.0) marker.

Contact title output

| ------------------------------------ |  |  |
| :--- | :--- | :--- |
| NTYPE | 1 | entity type $=90002$ |
| NUMCON | 1 | number of contacts |

For NUMCON contacts:

| IDC | 1 | contact id |
| :--- | :--- | :--- |
| CTITLE | 18 | Contact title (72 characters) |

Header output

| ------------------------------------ |  |  |
| :--- | :--- | :--- |
| NTYPE | 1 | entity type $=90000$ |
| HEAD | 18 | Header title $(72$ characters $)$ |

ICFD Part title output

| Value | Length | Description |
| :--- | :--- | :--- |
| ---------------------------- |  |  |
| NTYPE | 1 | entity type $=90020$ |
| NUMPROP | 1 | number of parts |

CESE: mechanics solid surface part title output

| Value | Length | Description |
| :--- | :--- | :--- |
| $----------------------\quad$ entity type $=90021$ |  |  |
| NTYPE | 1 | number of parts |

For NUMPROP parts:

| IDP | 1 | part id |
| :--- | :--- | :--- |
| PTITLE | 18 | Part title (72 characters) |

The d3prop data is written to the d3plot file only if it is requested.
D3PROP output
Values Length Description

| NTYPE | 1 | entity type $=900100$ |
| :--- | :--- | :--- |
| NLINE | 1 | number of keyword lines |

For NLINE keyword lines:
KEYWORD 20 keyword line (80 characters)

## DESCRIPTION OF BINARY FILE TYPES

| Control word 11 |  |
| :--- | :--- |
| File type:  <br> $1=\mathrm{d} 3$ plot plot file of model and state data <br> $2=\mathrm{d} 3 \mathrm{drlf}$ plot file of model and state data from a dynamic relaxation analysis <br> $3=\mathrm{d} 3$ thdt time history plot file for a set of nodes and elements <br> $4=$ intfor plot file of contact interfaces <br> $5=\mathrm{d} 3$ part plot file of model and state data for a set of parts <br> $6=$ blstfor plot file for a blast wave analysis <br> $7=\mathrm{d} 3 \mathrm{cpm}$  <br> $8=\mathrm{d} 3$ ale  <br> or flot file for ale fluid-structure interface  <br> $11=\mathrm{d} 3$ eigv  <br> $12=\mathrm{plot}$ file for an eigen value analysis  <br> $13=\mathrm{d} 3$ iter  |  |

$21=\mathrm{d} 3$ ssd $\quad$ plot file for steady state dynamic response.
$22=\mathrm{d} 3 \mathrm{spcm} \quad$ plot file for response spectrum analysis.
$23=\mathrm{d} 3$ psd plot file for power spectral density of response, in random vibration.
$24=\mathrm{d} 3 \mathrm{rms} \quad$ plot file for root mean square of response, in random vibration.
$25=\mathrm{d} 3 \mathrm{ftg} \quad$ plot file for random fatigue analysis.
26=d3acs plot file for frequency domain acoustic FEM analysis

## EXTRA DATA TYPES (Output for Multi-Solver Analysis)

If NCFDV1 $=67108864$, then NCFDV2 will be the number of additional datasets from different solver-mesh combinations that are included in the d3plot file. One of each of the solver-mesh combinations listed below can be among the NCFDV2 datasets. Currently defined solver-mesh combinations follow.

For the following domain, the mesh can be completely different for each output state, so no mesh is output in this control block.

```
solver and domain ID: PFEM_IF
number of volume vars output: nvolvar_pfem
first volume variable ID: ID I
last volume variable ID: ID nvolvar_pfem
number of PFEM parts nPFEM_parts
first internal part ID: partI\overline{D 1}
last internal part ID: partID nPFEM_parts
first user part ID:
last user part ID:
user_partID \overline{1}
user_partID nPFEM_parts
```

For the following domain, the mesh can be completely different for each output state, so no mesh is output in this control block.

```
solver and domain ID: PFEM_IF_SURFACE
number of surface vars output: nsurfvar_pfem
first surface variable ID: ID I
last surface variable ID: ID nsurfvar_pfem
number of PFEM surface parts nPFEM_surfparts
first internal part ID: partID 1
last internal part ID: partID nPFEM surfparts
first user part ID: user_surfpart\overline{ID }1
last user part ID: user_surfpartID nPFEM_surfparts
```

For the following domain, the mesh can be completely different for each output state, so no mesh is output in this control block.

```
solver and domain ID: CESE
number of volume vars output: nvolvar_cese
first volume variable ID: ID 1
last volume variable ID: ID nvolvar_cese
number of CESE parts nCESE_parts
first internal part ID: partID 1
```

```
last internal part ID:
first user part ID:
last user part ID: user_partID nCESE_parts
```

For the following domain, the mesh can be completely different for each output state, so no mesh is output in this control block.

```
solver and domain ID: CESE_SURFACE
number of surface vars output: nsurfvar_cese
first surface variable ID: ID 1
last surface variable ID: ID nsurfvar_cese
number of CESE surface parts nCESE surfparts
first internal part ID: partI\overline{D }1
last internal part ID: partID nCESE_surfparts
first user part ID: user_surfpart\overline{ID 1}
last user part ID: user_surfpartID nCESE_surfparts
```

For the following domain, the mesh can be completely different for each output state, so no mesh is output in this control block.

```
solver and domain ID:
number of volume vars output: nvolvar_cese
first volume variable ID: ID 1
last volume variable ID: ID nvolvar_cese
number of CESE parts nCESE_parts
first internal part ID: partI\overline{D 1}
last internal part ID: partID nCESE_parts
first user part ID: user_partID \overline{1}
...
last user part ID: user_partID nCESE_parts
```

For the following domain, the mesh can be completely different for each output state, so no mesh is output in this control block.

```
solver and domain ID: CESE2D_SURFACE
number of surface vars output: nsurfvar_cese
first surface variable ID: ID 1
last surface variable ID: ID nsurfvar_cese
number of CESE surface parts nCESE surfparts
first internal part ID: partI\overline{D 1}
last internal part ID: partID nCESE_surfparts
first user part ID: user_surfpart\overline{ID 1}
last user part ID: user_surfpartID nCESE_surfparts
```

For the following domain, the mesh can be completely different for each output state, so no mesh is output in this control block.

```
solver and domain ID:
number of volume vars output: nvolvar_cese
first volume variable ID: ID 1
last volume variable ID: ID nvolvar_cese
number of CESE parts nCESE_parts
first internal part ID: partI\overline{D 1}
last internal part ID: partID nCESE_parts
first user part ID: user_partID \overline{1}
last user part ID: user_partID nCESE_parts
```

For the following domain, the mesh can be completely different for each output state, so no mesh is output in this control block.

```
solver and domain ID: CESE2DAXI_SURFACE
number of surface vars output: nsurfvar_cese
first surface variable ID: ID 1
last surface variable ID: ID nsurfvar_cese
number of CESE surface parts nCESE surfparts
first internal part ID: partI\overline{D I}
last internal part ID: partID nCESE_surfparts
first user part ID: user_surfpart\overline{ID }1
last user part ID: user_surfpartID nCESE_surfparts
```

In this domain, the variables are defined at the element centroid.

```
solver and domain ID: CESE CFD ELEMENT
size of each variable component: nume\overline{lh_cèse}
number of nodes:
number of elements:
user node numbers:
array of nodal coordinates:
element connectivity:
number of output vars:
first variable ID:
last variable ID:
number of CESE parts
first internal part ID:
last internal part ID:
first user part ID:
last user part ID:
user element number:
...
user element number:
```

```
numnp_\overline{cese}
```

numnp_\overline{cese}
numelh_cese
numelh_cese
nodes cese cfd(numnp cese)
nodes cese cfd(numnp cese)
x_ces\overline{e}cfd\overline{(3, numnp \overline{cese)}}\mathbf{(})
x_ces\overline{e}cfd\overline{(3, numnp \overline{cese)}}\mathbf{(})
ix8_cese_cfd(9, numelh_cese)
ix8_cese_cfd(9, numelh_cese)
nv_cese_cfd_ele
nv_cese_cfd_ele
ID 1
ID 1
ID nv_cese_cfd_ele
ID nv_cese_cfd_ele
ncese_parts
ncese_parts
partI\overline{D 1}
partI\overline{D 1}
partID ncese_parts
partID ncese_parts
user_partID \overline{1}
user_partID \overline{1}
user_partID ncese_parts
user_partID ncese_parts
for the first CESE element
for the first CESE element
for the last CESE element

```
for the last CESE element
```

In this domain, the variables are defined at the surface element centroid.

```
solver and domain ID: CESE SURFACE CFD ELEMENT
size of each variable component: nume\overline{l}
number of nodes:
number of elements:
user node numbers:
array of nodal coordinates:
element connectivity:
number of output vars:
first variable ID:
last variable ID:
number of CESE surface parts
first internal part ID:
last internal part ID:
first user part ID:
last user part ID: user_surfpartID ncese_surfparts
user element number: for the first CESE surface element
user element number:
```

In this domain, the variables are defined at the 2D element centroid.

```
solver and domain ID: CESE2D_CFD_ELEMENT
size of each variable component: numel2\overline{d_cese}
number of nodes: numnp cèse
number of elements: numel\overline{2}d_cese
user node numbers:
element connectivity:
number of output vars:
first variable ID:
last variable ID:
number of CESE parts
first internal part ID:
...
last internal part ID:
first user part ID:
last user part ID:
user element number:
user element number:
```

```
nodes_cese_cfd(numnp_cese)
```

```
nodes_cese_cfd(numnp_cese)
```




```
i\overline{x}4 ce\overline{se_cfd(5, numēl2d_cese)}
```

i\overline{x}4 ce\overline{se_cfd(5, numēl2d_cese)}
nv_cese_cfd_ele
nv_cese_cfd_ele
ID }\mp@subsup{}{}{-
ID }\mp@subsup{}{}{-
ID nv_cese_cfd_ele
ID nv_cese_cfd_ele
ncese_part\overline{s}
ncese_part\overline{s}
partI\overline{D}1
partI\overline{D}1
partID ncese_parts
partID ncese_parts
user_partID \overline{1}
user_partID \overline{1}
user_partID ncese_parts
user_partID ncese_parts
for the first 2D \overline{CESE element}
for the first 2D \overline{CESE element}
for the last 2D CESE element

```
for the last 2D CESE element
```

In this domain, the variables are defined at the 2 D surface element centroid.

```
solver and domain ID: CESE2D_SURFACE_CFD_ELEMENT
size of each variable component: numsur\overline{f}2Dele_cèse
number of nodes: numnp_cese
number of elements: numsurff2Dele_cese
user node numbers: nodes_cese_c\overline{f}d(numnp_cese)
array of nodal coordinates: x_ces\overline{e}_cfd\overline{(3, numnp_\overline{c}}\mathbf{~}=se)
element connectivity:
number of output vars:
first variable ID:
i\overline{x}2_ce\overline{se_cfd(3, numsurf2Dele_cese)}
nv_\overline{cese_\overline{cfd_ele}}\mathbf{}\mathrm{ - }
ID-1
```

```
last variable ID:
number of CESE surface parts
first internal part ID:
last internal part ID:
first user part ID:
..
last user part ID:
user element number:
user element number:
```

```
ID nv cese cfd ele
ncese_surfp
partID 1
partID ncese_surfparts
user_surfpartID 1
user surfpartID ncese surfparts
for the first CESE surface element
for the last CESE surface element
```

In this domain, the variables are defined at the 2 D axisymmetric element centroid.

```
solver and domain ID: CESE2DAXI_CFD_ELEMENT
size of each variable component: numel2daxi_cese
number of nodes: numnp cese
number of elements: numel\overline{2}daxi_cese
user node numbers: nodes_cese_cfd(numnp_cese)
array of nodal coordinates: x_ces\overline{e}_cfd(3, numnp_\overline{cese)}
element connectivity:
number of output vars:
first variable ID:
last variable ID:
number of CESE parts
first internal part ID:
last internal part ID:
first user part ID:
last user part ID:
user element number:
user element number:
```

```
i\overline{x}4_ce\overline{se_cfd(5, numèl2daxi_cese)}
```

i\overline{x}4_ce\overline{se_cfd(5, numèl2daxi_cese)}
nv_\overline{cese_\overline{cfd_ele}}\mathbf{}\mathrm{ - }
nv_\overline{cese_\overline{cfd_ele}}\mathbf{}\mathrm{ - }
ID}\mp@subsup{}{}{-1
ID}\mp@subsup{}{}{-1
ID nv_cese_cfd_ele
ID nv_cese_cfd_ele
ncese_parts
ncese_parts
partI\overline{D}1
partI\overline{D}1
partID ncese_parts
partID ncese_parts
user_partID \overline{1}
user_partID \overline{1}
user_partID ncese_parts
user_partID ncese_parts
for the first 2D \overline{CESE element}
for the first 2D \overline{CESE element}
for the last 2D CESE element

```
for the last 2D CESE element
```

In this domain, the variables are defined at the 2D axisymmetric surface element centroid.

```
solver and domain ID:
size of each variable component:
number of nodes:
number of elements:
user node numbers:
array of nodal coordinates:
element connectivity:
number of output vars:
first variable ID:
last variable ID:
number of CESE surface parts
first internal part ID:
last internal part ID:
first user part ID:
last user part ID:
user element number:
...
user element number:
```

```
CESE2DAXI_SURFACE_CFD_ELEMENT
```

CESE2DAXI_SURFACE_CFD_ELEMENT

```
numsurfel\overline{2daxi_cese}
```

numsurfel\overline{2daxi_cese}
numnp_cese
numnp_cese
numsurfel2daxi cese
numsurfel2daxi cese
nodes_cese_cfd(numnp_cese)
nodes_cese_cfd(numnp_cese)
x_cesè_cfd\overline{(3, numnp_cese)}
x_cesè_cfd\overline{(3, numnp_cese)}
ix2_cese_cfd(3, numsurfel2daxi_cese)
ix2_cese_cfd(3, numsurfel2daxi_cese)
nv_cese_cfd_ele
nv_cese_cfd_ele
ID 1
ID 1
ID nv_cese_cfd_ele
ID nv_cese_cfd_ele
ncese-surf\overline{p}art\overline{s}
ncese-surf\overline{p}art\overline{s}
partI\overline{D 1}
partI\overline{D 1}
partID ncese_surfparts
partID ncese_surfparts
user_surfpar\overline{tID 1}
user_surfpar\overline{tID 1}
user surfpartID ncese surfparts
user surfpartID ncese surfparts
for The first CESE suŕface element
for The first CESE suŕface element
for the last CESE surface element

```
for the last CESE surface element
```

For the following domain, the mesh can be completely different for each output state, so no mesh is output in this control block.

```
solver and domain ID: MECH_SURFACE
number of surface vars output: nsurfvar_mech
first surface variable ID: ID 1
last surface variable ID: ID nsurfvar_mech
number of MECH surface parts nMECH surfparts
first internal part ID: partID}
last internal part ID: partID nMECH_surfparts
first user part ID: user_surfpart\overline{ID}1
last user part ID: user_surfpartID nMECH_surfparts
```

For the following domain, the mesh can be completely different for each output state, so no mesh is output in this control block.

```
solver and domain ID: MECH2D_SURFACE
number of surface vars output: nsurfvar_mech
first surface variable ID: ID I
last surface variable ID: ID nsurfvar_mech
number of MECH 2D surface parts nMECH2D_surfparts
first internal part ID: partID 1
last internal part ID: partID nMECH2D_surfparts
first user part ID: user_surfpartID}
last user part ID: user_surfpartID nMECH2D_surfparts
```

In this domain, the variables are defined on structural solid elements.

```
solver and domain ID: EM_FEMSTER_SOLID_INTEG_PTS
size of each variable component: ni\overline{p}_solid_\overline{em * nūmelh}
number of nodes:
numnp
number of elements:
numelh
number of integration points: nip_solid_em
number of output vars: nv_em_solīd_integ
first variable ID: ID``
last variable ID: ID nv_em_solid_integ
```

In this domain, the variables are defined on structural thick shell elements.

```
solver and domain ID: EM_FEMSTER_TSHELL_INTEG_PTS
size of each variable component: ni\overline{p}_tshell_em * numelt
number of nodes: num\overline{n}
number of elements: numelt
number of integration points: nip_tshell_em
```

```
number of output vars:
first variable ID:
last variable ID: ID nv_em_tshell_integ
```

In this domain, the variables are defined on structural thin shell elements.

```
solver and domain ID:
size of each variable component: ni\overline{p}_shell_em * numels
number of nodes:
number of elements:
number of integration points: nip_shell_em
number of output vars: nv_em_shell_integ
first variable ID: ID 1
last variable ID: ID nv_em_shell_integ
EM_FEMSTER_SHELL_INTEG_PTS
numnp
numels
```

In this domain, the variables are defined at the centroids of structural solid elements.

```
solver and domain ID: EM_FEMSTER_SOLID_CENTROID
size of each variable component: numelh
number of nodes: numnp
number of elements: numelh
number of output vars: nv_em_solid_cent
first variable ID: ID }\mp@subsup{}{}{-
last variable ID: ID nv_em_solid_cent
```

In this domain, the variables are defined at the centroids of structural thick shell elements.

```
solver and domain ID: EM_FEMSTER_TSHELL_CENTROID
size of each variable component: numelt
number of nodes: numnp
number of elements: numelt
number of output vars: nv_em_tshell_cent
first variable ID: ID }\mp@subsup{}{}{-
last variable ID: ID nv_em_tshell_cent
```

In this domain, the variables are defined at the centroids of structural thin shell elements.

```
solver and domain ID: EM_FEMSTER_SHELL_CENTROID
size of each variable component: numels
number of nodes: numnp
number of elements: numels
number of output vars: nv_em_shell_cent
first variable ID: ID 1
last variable ID: ID nv_em_shell_cent
```

In this domain, the variables are defined at the mesh nodes.

```
solver and domain ID:
Solver and domain ID: component: nip__air_em * * numelh_air_em
number of nodes: numn̄p_aír_em
number of elements:
    numel\overline{h_air_em}
number of integration points: nip_air_em
```

```
user node numbers: nodes air em(numnp air em)
array of nodal coordinates: x_air_em(\overline{3},numnp_air_e\overline{m})
element connectivity:
number of output vars:
first variable ID:
last variable ID: ID nv_em_air_integ
array of nodal coordinates: x_air_em(\overline{3},numnp_air_em)
i\overline{x}8_a\overline{ir_em(8, numelh_air_em)}
nv_èm_aír_integ
ID 1
```

In this domain, the variables are defined at the nodes of the implied rectangular mesh.

In this domain, the variables are defined on faces of structural elements.

```
solver and domain ID: RECT_AIR_EM_NODE
```

solver and domain ID: RECT_AIR_EM_NODE
size of each variable component: nx_rect_\overline{air_em * ny_rect_air_em}
size of each variable component: nx_rect_\overline{air_em * ny_rect_air_em}
* \overline{nz_re\overline{c}t_a\overline{ir_em}}\mathbf{|}=\mp@code{m}
* \overline{nz_re\overline{c}t_a\overline{ir_em}}\mathbf{|}=\mp@code{m}
number of x nodes: nx_rect_air_em
number of x nodes: nx_rect_air_em
number of y nodes: ny_rect_air_em
number of y nodes: ny_rect_air_em
number of z nodes: nz_rect_air_em
number of z nodes: nz_rect_air_em
minimum x coordinate: xmín_re\overline{c}t_a\overline{ir_em}
minimum x coordinate: xmín_re\overline{c}t_a\overline{ir_em}
minimum y coordinate: ymin_rect_air_em
minimum y coordinate: ymin_rect_air_em
minimum z coordinate: zmin_rect_air_em
minimum z coordinate: zmin_rect_air_em
maximum x coordinate: xmax_rect_air_em
maximum x coordinate: xmax_rect_air_em
maximum y coordinate: ymax_rect_air_em
maximum y coordinate: ymax_rect_air_em
maximum z coordinate: zmax_rect_air_em
maximum z coordinate: zmax_rect_air_em
number of output vars: nv_em_air_nd
number of output vars: nv_em_air_nd
first variable ID: ID`1 first variable ID: ID`1
last variable ID: ID nv_em_air_nd
last variable ID: ID nv_em_air_nd
last variable ID: ID nv_em_air_nd

```
last variable ID: ID nv_em_air_nd
```

```
solver and domain ID: EM_FEMSTER_BEM
```

solver and domain ID: EM_FEMSTER_BEM
size of each variable component: nip}_bem_em * nfaces_bem_em
size of each variable component: nip}_bem_em * nfaces_bem_em
number of nodes: numñp_b\overline{em_em}
number of nodes: numñp_b\overline{em_em}
number of elements: nfaces_bem_em
number of elements: nfaces_bem_em
number of integration points: nip_bem_em
number of integration points: nip_bem_em
number of BEM parts: em_numPartBem
number of BEM parts: em_numPartBem
flag for BEM mesh: nBEMflag
flag for BEM mesh: nBEMflag
(first bit =0 if no motion,=1 if motion)
(first bit =0 if no motion,=1 if motion)
(second bit =0 if no edge domain,=1 if edge domain)
(second bit =0 if no edge domain,=1 if edge domain)
(third bit =0 if no node domain,=1 if node domain)
(third bit =0 if no node domain,=1 if node domain)
user node numbers: nodes_bem_em(numnp_bem_em)
user node numbers: nodes_bem_em(numnp_bem_em)
array of nodal coordinates:
array of nodal coordinates:
element connectivity:
element connectivity:
x_bem_em(\overline{3}, numnp_\overline{bem_}\overline{e}m)
x_bem_em(\overline{3}, numnp_\overline{bem_}\overline{e}m)
i\overline{x}4_b\overline{em_em(5, nfa\overline{ces_\overline{b}em_em)}}\mathbf{(})
i\overline{x}4_b\overline{em_em(5, nfa\overline{ces_\overline{b}em_em)}}\mathbf{(})
if (second bit(nBEMflag) = 1)
if (second bit(nBEMflag) = 1)
number of edges
number of edges
number of edge domains
number of edge domains
node edge connectivity
node edge connectivity
number of edges per domain
number of edges per domain
edge domain list
edge domain list
endif
endif
if (third bit(nBEMflag) = 1)
if (third bit(nBEMflag) = 1)
node element connectivity
node element connectivity
number of node domains
number of node domains
nodex4_bem_em(5, nfaces_bem_em)
nodex4_bem_em(5, nfaces_bem_em)
nnodedōmain__bem_em
nnodedōmain__bem_em
node domain array
node domain array
nedges bem em
nedges bem em
nedgedōmain_bem_em
nedgedōmain_bem_em
edgex2_bem_\overline{em(3,}\mathrm{ , nedges_bem_em)}
edgex2_bem_\overline{em(3,}\mathrm{ , nedges_bem_em)}
(interñal node1,interna\overline{l}}\mathrm{ no就2,partId)
(interñal node1,interna\overline{l}}\mathrm{ no就2,partId)
numEdgesPerDomain(nedgedomain_bem_em)
numEdgesPerDomain(nedgedomain_bem_em)
edgeDomainList(sum(numEdgePer\overline{D}omain(i)))
edgeDomainList(sum(numEdgePer\overline{D}omain(i)))
endif
endif
nodeDomain(\

```
nodeDomain(\
```

```
number of output vars:
first variable ID:
last variable ID: ID nv_em_bem_integ
```

In this domain, the variables are defined at the nodes of the mechanics structural mesh.

```
solver and domain ID: EM_FEMSTER_NODE
number of nodes: num}n
size of each variable component: numnp_em
node list (internal #s): numnp_em
number of output vars: nv_no\overline{de_em}
first variable ID:
last variable ID: ID nv_node_em
```

In this domain, the variables are defined at the particle positions.

```
solver and domain ID:
number of output vars:
first variable ID:
last variable ID:
```

```
STOCHASTIC_PARTICLES
n_prtcl_vars
ID 1
ID n_prtcl_vars
```

Notes:
The variable IDs are grouped into three groups:

1) D3PL_FIRST_SCALAR_ID $<=$ ID $<$ D3PL_FIRST_VECTOR_ID are scalar variables
2) D3PL_FIRST_VECTOR_ID $<=$ ID $<$ D3PL_FIRST_TENSOR_ID are vector variables (3 components per entry)
3) D3PL_FIRST_TENSOR_ID $<=$ ID $<$ D3PL_END_IDS are symmetric tensor variables ( 6 component per entry)

When a number of integration points are specified, it is assumed that they are distributed at the Gauss points of the given element type based upon how many are output. That is, for shell or face elements, 4 output points would imply the $2 \times 2$ Gauss points are used, while 9 output points would imply the $3 \times 3$ Gauss points are used, and so forth. Similarly, for volume elements, 8 utput points would imply the $2 \times 2 \times 2$ Gauss points are used, while 27 output points would imply he $3 \times 3 \times 3$ Gauss points are used, and so forth.

## STATE DATA - d3plot and d3part

The state data has three parts:

- Time word and global data
- Node data
- Element data for solids, shell, and beams, respectively

| VALUE | LENGTH | DESCRIPTION |
| :---: | :---: | :---: |
| TIME | 1 | Time word |
| GLOBAL | NGLBV | Global variables for this state. LS-DYNA Global Variables: <br> KE, IE, TE, X, Y, and $Z$ velocity <br> IE for each material <br> KE for each material <br> $\mathrm{X}, \mathrm{Y}$, and Z velocity for part 1 <br> ... <br> $\mathrm{X}, \mathrm{Y}$, and Z velocity for part n <br> Mass for each part <br> Hourglass energy for each part <br> Force for each rigid wall <br> Xyz position of wall (1s971) <br> $=6+7$ * (NUMMAT8 + NUMMAT2 + <br> NUMMAT4 + NUMMATT+NUMRBS) + N * <br> NUMRW, $\mathrm{N}=1$, for ls-dyna(ls971) $\mathrm{N}=4$ |
| NODEDATA | NND | Total nodal values for state. <br> For LS-DYNA3D and LS-DYNA <br> $\mathrm{IT}=1$, node temperatures only, $\mathrm{N}=0$ <br> $\mathrm{IT}=2$, node temperature and node flux, $\mathrm{N}=2$ <br> $\mathrm{IT}=3,3$ temperature per node and node flux, $\mathrm{N}=3$ <br> Temperature for shell node at inner, middle and outer layer, inner array, middle array, outer array. IT $/ 10=1$, mass scaling value at node. $\mathrm{N}+=1$ $=((\mathrm{IT}+\mathrm{N})+\mathrm{NDIM} *(\mathrm{IU}+\mathrm{IV}+\mathrm{IA})) * \mathrm{NUMNP}$ where IT=temperature flag, $\mathrm{IU}=$ coordinates flag, $\mathrm{IV}=$ velocities flag, and $\mathrm{IA}=$ accelerations flag. |
| THERMDATA |  | If NT3D $>0$, NT3D * NEL8 of thermal element data. |
| CFDDATA | CFD | Bit flag: NCFDV1, bits from right to left eg, Pressure, Resultant Vorticity, and Density NCFDV1 $=2+32+1024=1058$ <br> 2 Pressure <br> 3 X Vorticity <br> 4 Y Vorticity <br> 5 Z Vorticity |

6 Resultant Vorticity
7 Enstrophy
8 Helicity
9 Stream Function
10 Enthalpy
11 Density
12 Turbulent KE
13 Dissipation
14-20 Eddy Viscosity
Bit flag: NCFDV2
2-11 Species 1 through 10

## NOTE: This CFDDATA is no longer output by ls-dyna.

ELEMDATA ENN | Total element data for state. |
| :--- |
| $=$ NEL8*NV3D+NELT*NV3DT+NEL2*NV1D+ |
| NEL4*NV2D+NMSPH*NUM_SPH_VARS |
| The organization of the element data for each |
| element type is described below. The data for the |
| solid elements (7 values/element) is printed first, |
| followed by the data for the beam elements ( 6 |
| values/element), and then the data for the shell |
| elements (typical 33 or 45 values/element |
| depending on whether the strains are included). |

This state data is repeated for each state in the database.
Element data is defined at the integration points within the element. Contour and fringe plots require that the data be extrapolated to or averaged at the nodal points. In LS_PREPOST the element values are averaged at the nodes or optionally extrapolated to the nodes. Element strains are not output by default, these are only output for solids, shell, and thick shell when *DATABASE_EXTENT_BINARY, STRFLG=1

SOLID ELEMENTS - 8 node Hexahedron, other solid elements like wedge, pyramid, and tetrahedron are identified by repeated final connectivities. Eg tetrahedron $=1,2,3,4,4,4,4,4$

The database for solid elements consists of 7+NEIPH values per element. NEIPH extra values are defined if and only if NEIPH is greater than zero or if the model is an ALE analysis. If strain components are output, then the last 6 neiph values are true strains: ex, ey, ez, exy, eyz, exz, in the global system.
They are:

1. Sigma-x (true stress in the global system)
2. Sigma-y
3. Sigma-z
4. Sigma-xy
5. Sigma-yz
6. Sigma-zx
7. Effective plastic strain or material dependent variable
8. First extra value (if NEIPH $>0$ )
9. Second extra value (if NEIPH $>1$ )
10. Etc. until NEIPH extra values are defined if ISTRN=1

7+NEIPH-5. Epsilon-x
7+NEIPH-4. Epsilon-y
7+NEIPH-3. Epsilon-z
7+NEIPH-2. Epsilon-xy
7+NEIPH-1. Epsilon-yz
7+NEIPH. Epsilon-zx

For thick shell elements the database contains NV3DT $=$ MAXINT $*(6 * \operatorname{IOSHL}(1)+$ IOSHL(2) + NEIPS) +12 * ISTRN values per element. Three sets of global stresses are always put into the database for each thick shell and are located at the mid surface, the inner integration point surface, and the outer integration point surface, respectively. If one integration point is used the same through the thickness stress state is outputted three times. If two integration points are used then the mid surface value is taken as the average value. The inner values of the stress are always set to the values at the innermost integration point and likewise for outer values. If the integration point does not lie at the center, ie, an even number of integration points through the thickness, a value is computed that is an average of the two integration point nearest the mid surface.
The IOSHL flags indicate which shell element data is included which is suppressed.
The flags are set in ls-dyna by *DATABASE_EXTENT_BINARY, SIGFLG, EPSFLG, RLFLG, and ENGFLG

The ordering of the data follows:

1. Sigma-x (mid surface true stress in global system)
2. Sigma-y
3. Sigma-z
4. Sigma-xy
5. Sigma-yz
6. Sigma-zx
7. Effective plastic strain or material dependent variable
*. Define NEIPS additional history values here for mid surface
8. Sigma-x (inner surface true stress in global system)
9. Sigma-y
10. Sigma-z
11. Sigma-xy
12. Sigma-yz
13. Sigma-zx
14. Effective plastic strain or material dependent variable
*. Define NEIPS additional history values here for inner surface
15. Sigma-x (outer surface true stress in global system)
16. Sigma-y
17. Sigma-z
18. Sigma-xy
19. Sigma-yz
20. Sigma-zx
21. Effective plastic strain or material dependent variable
*. Define NEIPS additional history values here for outer surface
*. If ISTRN=1, then define strain components Epsilon ( $\mathbf{x}, \mathbf{y}, \mathbf{z}, \mathbf{x y}, \mathbf{y z}, \mathbf{z x}$ ) here for inner surface and outer surface

If MAXINT > 3 then define an additional (MAXINT-3 $)^{*}(6 * \operatorname{IOSHL}(1)+1 * \operatorname{IOSHL}(2)+$ NEIPS $)$ quantities here.

For beam elements the database contains NV1D $=6$ values per element. They are:

1. Axial force
2. S shear resultant
3. T shear resultant
4. $\quad \mathrm{S}$ bending moment
5. T bending moment
6. Torsional resultant

If there are values output at beam integration points, then NV1D $=6+5 *$ BEAMIP

1. RS shear stress
2. TR shear stress
3. Axial stress
4. Plastic strain
5. Axial strain

BEAMIP is set in *DATABASE_EXTENT_BINARY

If beam material history variables are requested, NEIPB*(3+BEAMIP) number of values are written after the previous data.
Average, min and max, computed from all integration points by LS-DYNA, for NEIPB history variables are written. Then followed by NEIPB history variables for BEAMIP number of integration points.
For example, if $\mathrm{NEIPB}=2$ and $\mathrm{BEAMIP}=3$ the data comes in this order:
$\left\{h v 1 \_a v g\right.$, hv2, avg $\},\left\{h v 1 \_m i n, ~ h v 2 \_\min \right\},\left\{h v 1 \_\max , \mathrm{hv} 2 \_\max \right\},\left\{h v 1 \_i p 1, \mathrm{hv} 2 \_i p 1\right\}$, $\left\{h v 1 \_i p 2, h v 2 \_i p 2\right],\left\{h v 1 \_i p 3, h v 2 \_i p 3\right\}$.

For shell elements the database contains NV2D values, where:
$\mathrm{NV} 2 \mathrm{D}=$ MAXINT* $(6 * \operatorname{IOSHL}(1)+1 * \operatorname{IOSHL}(2)+\mathrm{NEIPS})+8 * \operatorname{IOSHL}(3)+4 * \operatorname{IOSHL}(4)+$ $12 *$ ISTRN values per deformable element. If MATTYP $=1$ and $\operatorname{IRBTYP}(\mathrm{I})=20$, where $\mathrm{I}=$ internal element number, then the material is rigid and the compressed database contains no data for the element. If the minimum value of $M A X I N T$ is used, i.e., $=3$, the stresses are typically located at the mid surface, the inner surface, and the outer surface, respectively. If one integration point is used the stress is written three times. If two integration points are used then the mid surface value is taken as the average value. The inner values of the stress are always set to the values at the innermost integration point and likewise for outer values. If no integration point lies at the center, i.e., an even number of integration points through the thickness, a value is computed that is an average of the two integration point lying nearest the mid surface.

The ordering of the data follows:

1. Sigma-x (mid surface true stress in global system)
2. Sigma-y
3. Sigma-z
4. Sigma-x
5. Sigma-yz
6. Sigma-zx
7. Effective plastic strain or material dependent variable
*. Define NEIPS additional history values here for mid surface
8. Sigma-x (inner surface true stress in global system)
9. Sigma-y
10. Sigma-z
11. Sigma-xy
12. Sigma-yz
13. Sigma-zx
14. Effective plastic strain or material dependent variable

## *. Define NEIPS additional history values here for inner surface

15. Sigma-x (outer surface true stress in global system)
16. Sigma-y
17. Sigma-z
18. Sigma-xy
19. Sigma-yz
20. Sigma-zx
21. Effective plastic strain or material dependent variable
*. Define NEIPS additional history values here for outer surface
If MAXINT $>3$ then define an additional (MAXINT-3 )* (6*IOSHL(1) $+1 * \operatorname{IOSHL}(2)+$ $8 * \operatorname{IOSHL}(3)+4 * \operatorname{IOSHL}(4)+\mathrm{NEIPS})$ quantities here
22. Bending moment-Mx (local shell coordinate system)
23. Bending moment-My
24. Bending moment-Mxy
25. Shear resultant-Qx
26. Shear resultant-Qy
27. Normal resultant-Nx
28. Normal resultant-Ny
29. Normal resultant-Nxy
30. Thickness
31. Element dependent variable
32. Element dependent variable
33. Internal energy (if and only if ISTRN=0)

The following quantities are expected if and only if ISTRN=1
33. eps-x (inner surface strain in global system)
34. eps-y
35. eps-z
36. eps-xy
37. eps-yz
38. eps-zx
39. eps-x (outer surface strain in global system)
40. eps-y
41. eps-z
42. eps-xy
43. eps-yz
44. eps-zx
45. Internal energy (if and only if NV2D $>=45$ )

## ELEMENT DELETION OPTION

Skip this section if the word MAXINT is greater than or equal to zero, (MDLOPT $>=0$ ). If MDLOPT $=1$, then the list is equal to the number of nodal points (NUMNP) and contains a one if the node is visible and a zero if the node is not visible, (only used in vec-dyna3d). If MDLOPT=2, then the list equals the total number of elements (NEL8 + NELT + NEL4 + NEL2), in this order, and each value is set to the element material number or $=0$, if the element is deleted. All these numbers are output as floating point values and not integers.

## SMOOTH PARTICLE HYDRODYNNAMICS NODE/ELEMENT STATE DATA

This section is only output if NMSPH $>0$
For each SPH node the follow values are output:
NUM_SPH_DATA $=1+\sum$ isphfg(i), $i=2: 10$
Length of data $=$ NUM_SPH_DATA * NUMSPH
Material number, if $<=0$ then element is deleted.
Currently $\operatorname{isphfg}(1)=10$, ie number of sph data flags, this could be changed in the future.
If isphfg(2) $=1$, radius of particle influence
If isphfg $(3)=1$, pressure in particle
If isphfg(4) $=6$, stress components for particle, $s x, s y, s z, s x y, s y z, ~ s x z$
If isphfg $(5)=1$, plastic strain for particle
If isphfg(6) $=1$, density of particle material
If isphfg(7) $=1$, internal energy of particle
If isphfg $(8)=1$, number of particle neighbors
If isphfg $(9)=6$, strain components for particle, ex, ey, ez, exy, eyz, exz
If isphfg(10)=1, mass of element (ls971)

Note: it is possible a SPH element could be deleted, or be none active in the initial states, and become active in later states.

## PARTICLE STATE DATA (NPEFG > 0)

DES DATA - see description below for state data

STATE DATA
NPARTGAS blocks of NSTGEOM data to describe the state geometry for each bag:

1. number of active particles
2. current bag volume

## PARTICLE DATA

NVAR words of data output for each particle:

1. gas ID
2. chamber ID
3. leakage flag, 0 active, -1 fabric, -2 vent hole, -3 mistracked
4. mass
5. radius
6. spin energy
7. translational energy
8. distance from particle to nearest segment
9. x position
10. y position
11. z position
12. x velocity
13. y velocity
14. z velocity

## ROAD SURFACE MOTION

If NDIM $>5$ output rigid body displacement, $d x, d y, d z$ and velocity, $v x, v y, v z$ of each road surface.
Length of data $=6 *$ NSURF

## RIGID BODY MOTION DATA, NDIM=8,9

see DCOMP=5,6 in *DATABASE_EXTENT_BINARY
For each rigid body:
$\mathrm{X}, \mathrm{Y}, \mathrm{Z}$ position of geometric center
MXYZ, Rotation matrix from of principal axes, 3 direction cosines, 9 ( $3 \times 3$ ) values
VX, VY, VZ translational velocity of GC
RVX, RVY, RVZ, rotational velocity of GC
AX, AY, AZ, translational acceleration of GC
RAX, RAY, RAZ, rotational acceleration of GC

If NDIM $=9$ only $\mathrm{X}, \mathrm{Y}, \mathrm{Z}$ and MXYZ are output.

## EXTRA DATA (Multi-Solver Analysis)

If NCFDV1 $=67108864$, then the state data includes NCFDV2 additional datasets from solver-mesh combinations specified after the "User material, node, and element identification numbers" for the structural mesh.

State data of the first solver-mesh combination
State data of the last (NCFDV2-th) solver-mesh combination

When the state data comes from the PFEM_IF domain, then the mesh is output first, followed by the data. Currently, the mesh is entirely tetrahedral, but we anticipate users will also specify mixed meshes in the near future:

| size of each volume variable component: | nnpvol_pfem |
| :--- | :--- |
| number of volume nodes: | nnpvol_pfem |
| number of tetrahedral elements: | ntet_pfem |
| number of pyramid elements: | npyr_pfem |
| number of wedge elements: | nwdg_pfem |
| number of hexahedral elements: | nhex_pfem |
| user volume node numbers: | volnodes_pfem(nnpvol_pfem) |
| array of volume nodal coordinates: | xvol_pfem(3, nnpvol_pfem) |
| tetrahedral element connectivity: | ix4_pfem(5, ntet_pfem) |
| pyramid element connectivity: | ix5_pfem(6, npyr_pfem) |
| wedge element connectivity: | ix6_pfem(7, nwdg_pfem) |
| hexahedral element connectivity: | ix8_pfem(9, nhex_pfem) |
|  |  |
| data for 1 st volume variable | (size is nnpvol_pfem) |

data for nvolvar_pfem-th volume variable (size is nnpvol_pfem)

## Notes:

(1) the first four entries of ix $4 \_$pfem ( 5 , ntet $\_$pfem $)$for each element are the volume node numbers, while the last entry is the volume part number.
(2) the first five entries of ix5_pfem(6, npyr_pfem) for each element are the volume node numbers, while the last entry is the volume part number.
(3) the first six entries of ix6_pfem(7, nwdg_pfem) for each element are the volume node numbers, while the last entry is the volume part number.
(4) the first eight entries of ix8_pfem(9, nhex_pfem) for each element are the volume node numbers, while the last entry is the volume part number.

When the state data comes from the PFEM_IF_SURFACE domain, then the surface mesh is output first, followed by the data. Currently, the surface mesh is entirely triangular, but we anticipate users will also specify mixed triangle-quadrilateral meshes in the near future:

```
size of each surface variable component: nnpsurf_pfem
number of surface nodes: nnpsurf_pfem
number of surface elements: nelsurf_pfem
user surface node numbers: surfnodes_pfem(nnpsurf_pfem)
surface element connectivity: ixsurf_pfem(5, nelsurf_pfem)
data for 1st surface variable (size is nnpsurf_pfem)
data for nsurfvar_pfem-th surface variable (size is nnpsurf_pfem)
```


## Notes:

(1) the first four entries of ixsurf_pfem(5, nelsurf_pfem) for each surface element are the surface node numbers, while the last entry is the surface part number. The 3rd and 4th node numbers are the same for triangles.

When the state data comes from the CESE domain, then the mesh is output first, followed by the data:

```
size of each volume variable component: nele_cese
number of volume nodes:
number of tetrahedral elements:
number of pyramid elements:
number of wedge elements:
number of hexahedral elements:
user volume node numbers:
array of volume nodal coordinates:
tetrahedral element connectivity:
pyramid element connectivity:
nnpvol_cese
ntet_cese
npyr_cese
nwdg_cese
nhex_cese
volnodes_cese(nnpvol_cese)
xvol_cese(3, nnpvol_cese)
ix4_cese(5, ntet_cese)
ix5_cese(6, npyr_cese)
```

wedge element connectivity: hexahedral element connectivity:
data for 1st volume variable

```
    ix6_cese(7, nwdg_cese)
```

    ix8_cese(9, nhex_cese)
    (size is nele_cese)
data for nvolvar_cese-th volume variable (size is nele_cese)
Notes:
(1) nele_cese $=$ ntet_cese + npyr_cese + nwdg_cese + nhex_cese
(2) the first four entries of ix4_cese(5, ntet_cese) for each element are the volume node numbers, while the last entry is the volume part number.
(3) the first five entries of ix5_cese(6, npyr_cese) for each element are the volume node numbers, while the last entry is the volume part number.
(4) the first six entries of ix6_cese(7, nwdg_cese) for each element are the volume node numbers, while the last entry is the volume part number.
(5) the first eight entries of ix8_cese(9, nhex_cese) for each element are the volume node numbers, while the last entry is the volume part number.

When the state data comes from the CESE_SURFACE domain, then the surface mesh is output first, followed by the data. The user surface node numbers reference nodes from the CESE domain:
size of each surface variable component: nnpsurf_cese
number of surface nodes: nnpsurf_cese
number of surface elements: nelsurf_cese
user surface node numbers: surfnodes_cese(nnpsurf_cese)
surface element connectivity: ixsurf_cese(5, nelsurf_cese)
data for 1st surface variable (size is nnpsurf_cese)
data for nsurfvar_cese-th surface variable (size is nnpsurf_cese)

## Notes:

(1) the first four entries of ixsurf_cese(5, nelsurf_cese) for each surface element are the surface node numbers, while the last entry is the surface part number. The 3 rd and 4th node numbers are the same for triangles.

When the state data comes from the CESE2D domain, then the mesh is output first, followed by the data:
size of each volume variable component: nele2d_cese
number of volume nodes:
number of triangle elements:
number of quadrilateral elements:
nnp2d_cese
ntri_cese
nquad_cese
user volume node numbers:
array of volume nodal coordinates: triangle element connectivity: quadrilateral element connectivity:
data for 1st volume variable
volnodes_cese(nnp2d_cese)
xvol_cese(3, nnp2d_cese)
ix3_cese(4, ntri_cese)
ix4_cese(5, nquad_cese)
(size is nele2d_cese)
data for nvolvar_cese-th volume variable (size is nele2d_cese)
Notes:
(1) nele2d_cese $=$ ntri_cese + nquad_cese
(2) the first three entries of ix3_cese(4, ntri_cese) for each element are the volume node numbers, while the last entry is the volume part number.
(3) the first four entries of ix4_cese(5, nquad_cese) for each element are the volume node numbers, while the last entry is the volume part number.

When the state data comes from the CESE2D_SURFACE domain, then the surface mesh is output first, followed by the data. The user surface node numbers reference nodes from the CESE2D domain:
size of each surface variable component: nnpsurf2d_cese
number of surface nodes: nnpsurf2d_cese
number of surface elements: nsurfel2d_cese
user surface node numbers: surfnodes_-cese(nnpsurf2d_cese)
surface element connectivity: ixsurf_cese(3, nsurfel2d_cese)
data for 1st surface variable (size is nnpsurf2d_cese)
data for nsurfvar_cese-th surface variable (size is nnpsurf2d_cese)
Notes:
(1) the first two entries of ixsurf_cese(3, nsurfel2d_cese) for each surface element are the surface node numbers, while the last entry is the surface part number.

When the state data comes from the CESE2DAXI domain, then the mesh is output first, followed by the data:
size of each volume variable component: nele2daxi_cese number of volume nodes:
nnp2daxi_cese
number of triangle elements:
ntri_cese
number of quadrilateral elements:
user volume node numbers:
array of volume nodal coordinates:
triangle element connectivity:
nquad_cese
volnodes_cese(nnp2daxi_cese)
xvol_cese(3, nnp2daxi_cese)
ix3_cese(4, ntri_cese)
quadrilateral element connectivity: ix4_cese(5, nquad_cese)
data for 1st volume variable
(size is nele2daxi_cese) data for nvolvar_cese-th volume variable (size is nele2daxi_cese)

Notes:
(1) nele2daxi_cese $=$ ntri_cese + nquad_cese
(2) the first three entries of ix3_cese(4, ntri_cese) for each element are the volume node numbers, while the last entry is the volume part number.
(3) the first four entries of ix4_cese(5, nquad_cese) for each element are the volume node numbers, while the last entry is the volume part number.

When the state data comes from the CESE2DAXI_SURFACE domain, then the surface mesh is output first, followed by the data. The user surface node numbers reference nodes from the CESE2DAXI domain:
size of each surface variable component: nnpsurf2d_cese
number of surface nodes: nnpsurf2d_cese
number of surface elements: nsurfel2d_cese
user surface node numbers: surfnodes_cese(nnpsurf2d_cese)
surface element connectivity: ixsurf_cese(3, nsurfel2d_cese)
data for 1st surface variable (size is nnpsurf2d_cese)
data for nsurfvar_cese-th surface variable (size is nnpsurf2d_cese)
Notes:
(1) the first two entries of ixsurf_cese(3, nsurfel2d_cese) for each surface element are the surface node numbers, while the last entry is the surface part number.

When the state data comes from the MECH_SURFACE domain, then the surface mesh is output first, followed by the data. The surface node numbers reference nodes from the mechanics mesh:

```
size of each surface variable component: nnpsurf
number of surface nodes: nnpsurf
number of surface elements: nelsurf
surface node numbers (internal): surfnodes(nnpsurf)
surface element connectivity: ixsurf(5, nelsurf)
data for 1st surface variable (size is nnpsurf)
data for nsurfvar-th surface variable (size is nnpsurf)
```

Notes:
(1) the first four entries of ixsurf(5, nelsurf) for each surface element are the surface node numbers (internal), while the last entry is the surface part number. The 3rd and 4th node numbers are the same for triangles.

When the state data comes from the MECH2D_SURFACE domain, then the surface mesh is output first, followed by the data. The surface node numbers reference nodes from the mechanics mesh:
size of each surface variable component: nnpsurf2d
number of surface nodes: nnpsurf2d
number of surface elements: nsurfel2d
surface node numbers (internal): surfnodes(nnpsurf2d)
surface element connectivity:
$\operatorname{ixsurf}(3$, nsurfel2d)
data for 1st surface variable
(size is nnpsurf2d)
data for nsurfvar-th surface variable (size is nnpsurf2d)

## Notes:

(1) the first two entries of ixsurf( 3 , nsurfel2d) for each surface element are the surface node numbers (internal), while the last entry is the surface part number.

When the state data comes from the STOChASTIC_PARTICLES domain, then the

```
size of each variable component: n_particles
array of particle positions: x_particles(3, n_particles)
data for 1st output variable
data for n_prtcl_vars-th output variable
```

There will always be at least the following two variables output for each particle domain: PARTICLE_SIZES and PARTICLE_VELOCItIES. That is, n_prtcl_vars >= 2. For each particle, both the position and velocity are a 3-component vector.

## END OF FILE MARKER

Value $=-999999.0($ a floating point number $)$

## TIME HISTORY DATABASE (d3thdt)

There are three sections in the LS-DYNA time history database. The first used to contain 144 words of control information, but now depends upon the number of node and elements the user defines in LS-DYNA. The second contains geometric information including the nodal coordinates and element connectivities. The third section contains the results of the analysis at sequential output intervals for a subset of solids, beams, and shells. The output at a given time, called a state, contains a time word, global variables such as total energies and momenta, nodal data consisting of accelerations, velocities, and displacements, and finally element data is written that may include stresses and strains at integration points. The control information provides information on what is in the file and which database is contained.

## CONTROL DATA

| VALUE | \#WORDS | DISK <br> ADDRESS | DESCRIPTION |
| :--- | :--- | :--- | :--- |
| Title | 10 | 0 | Model Identification |
| Run time | 1 | 10 | Time in seconds since $00: 00: 00$ UTC, January 1, <br> 1970 |
| File type | 1 | 11 | $1=\mathrm{d} 3 \mathrm{plot}, 2=\mathrm{d} 3 \mathrm{drlf}, 3=\mathrm{d} 3 \mathrm{thdt}, 4=\mathrm{intfor}, 5=\mathrm{d} 3 \mathrm{part}$, <br> $6=\mathrm{blstfor} 7=,\mathrm{d} 3 \mathrm{cpm}, 8=\mathrm{d} 3 \mathrm{ale}, 11=\mathrm{d} 3 \mathrm{eigv}$, <br> $12=\mathrm{d} 3 \mathrm{mode}, 13=\mathrm{d} 3 \mathrm{iter}, 21=\mathrm{d} 3 \mathrm{ssd}, 22=\mathrm{d} 3 \mathrm{spcm}$, <br> $23=\mathrm{d} 3 \mathrm{psd}, 24=\mathrm{d} 3 \mathrm{rms}, 25=\mathrm{d} 3 \mathrm{ftg}, 26=\mathrm{d} 3 \mathrm{acs}$ |
| Source version | 1 | 12 | LS-DYNA version $* 1000000+\mathrm{svn}$ number |
| Release number | 1 | 13 | Release number in character*4 form <br> 50 for R5.0 <br> 511 c for R5.1.1c |
| Version | 1 | 14 | Code version, a real number, not integer |
| NDIM | 1 | 15 | Number of dimensions $(2$ or 3) is set to 4 if <br> element connectivies are unpacked in the LS- <br> DYNA/3D |
| NUMNP | 1 | 16 | Number of nodal points |
| ICODE | 1 | 17 | Flag to identify finite element code <br> $=2$ old DYNA3D, NIKE3D database <br> $=6$ new LS-NIKE3D, LS-DYNA/3D database |
| NGLBV | 1 | 18 | Number of global variables to be read in each <br> state |
| IT | 1 | 19 | Flag for temperatures <br> $=0$ none, |


| VALUE | \#WORDS | DISK <br> ADDRESS | DESCRIPTION |
| :---: | :---: | :---: | :---: |
|  |  |  | $=1$ read in a temperature for each node |
| IU | 1 | 20 | Flag for current geometry (=1) |
| IV | 1 | 21 | Flag for velocities (=1) |
| IA | 1 | 22 | Flag for accelerations ( $=1$ ) |
| NEL8 | 1 | 23 | Number of 8 node solid elements |
| NUMMAT8 | 1 | 24 | Number of materials (parts) used by the 8 node solids |
| NDS | 1 | 25 | Number of node blocks for plotting |
| NST | 1 | 26 | Number of element blocks for plotting. $=$ NSTH + NSTB + NSTS +NSTT |
| NV3D | 1 | 27 | Number of values in database for each solid element |
| NEL2 | 1 | 28 | Number of 2 node one-dimensional elements |
| NUMMAT2 | 1 | 29 | Number of materials (parts) used by the 2 node 1D elements |
| NV1D | 1 | 30 | Number of values in database for each 1D element |
| NEL4 | 1 | 31 | Number of four node two-dimensional elements |
| NUMMAT4 | 1 | 32 | Number of materials (parts) used by the 4 node 2D elements |
| NV2D | 1 | 33 | Number of values in database for each 2D element |
| NEIPH | 1 | 34 | Number of additional values per solid element to be written in the type 6 database $=\text { NEIPH-6*ISTRN }$ |
| NEIPS | 1 | 35 | Number of additional values per integration point to be written into the type 6 database for shell elements |
| MAXINT | 1 | 36 | Number of integration points dumped for each shell element |
| NMSPH | 1 | 37 | Number of SPH nodes |
| NGPSPH | 1 | 38 | Number of SPH materials |
| NARBS | 1 | 39 | Additional storage required for arbitrary node and element numbering in type 6 database |
| BLANK | 3 | 40 | Unused space |
| IOSHL(1) | 1 | 43 | Stress components flag (=1000 yes) |


| VALUE | \#WORDS | DISK <br> ADDRESS | DESCRIPTION |
| :---: | :---: | :---: | :---: |
| IOSHL(2) | 1 | 44 | Strain components, ISTRN ( $=1000$ yes) |
| IOSHL(3) | 1 | 45 | Shell force resultants ( $=1000$ yes) |
| IOSHL(4) | 1 | 46 | Shell thickness, energy +2 others ( $=1000$ yes) |
| BLANK | 1 | 47 | Unused space, ignore value |
| NCFDV1 | 1 | 48 | Bit flags for CFD nodal values |
| NCFDV2 | 1 | 49 | Further bit flags for CFD nodal values |
| BLANK | 7 | 50 | Unused space |
| EXTRA | 1 | 57 | Additional number of control words. If $>0$, there are EXTRA control words after the first 64 words. |
| NSTP | 1 | 58 | Number of SPH element blocks |
| IFLAGD | 1 | 59 | Number of node blocks + 1000 flag |
| NSTH | 1 | 60 | Number of solid element blocks |
| NSTB | 1 | 61 | Number of beam element blocks |
| NSTS | 1 | 62 | Number of shell element blocks |
| NSTT | 1 | 63 | Number of thick shell element block |
| NDSB | $2 * N D S$ | 64+EXTRA | Node blocks for which time histories are output. The locations $2 \mathrm{n}-1$, where $\mathrm{n}=1$ through NDS correspond to the first node in the block and locations 2 n correspond to the last node in the block. |
| NSTHB | $2 *$ NSTH | $\begin{aligned} & \text { 64+EXTRA+ } \\ & 2 * \text { NDS } \end{aligned}$ | Solid element blocks start and end numbers. These are defined in a similar manner to the nodal time history blocks |
| NSTBB | $2 *$ NSTB | $\begin{aligned} & \text { 64+EXTRA+ } \\ & \text { 2*NDS+ } \\ & 2 * \text { NSTH } \end{aligned}$ | Beam element block start and end numbers |
| NSTSB | $2 *$ NSTS | $\begin{aligned} & \text { 64+EXTRA+ }+ \\ & 2 * \text { NDS }+ \\ & 2 * \text { NSTH }+ \\ & 2 * \text { NSTB } \end{aligned}$ | Shell element block start and end numbers |
| NSTTB | $2 *$ NSTT | $\begin{aligned} & \text { 64+EXTRA+ }+ \\ & 2 * \text { NDS }+ \\ & 2 * \text { NSTH }+ \\ & 2 * \text { NSTB }+ \\ & 2 * \text { NSTS } \end{aligned}$ | Thick shell element block start and end numbers |
| NSTPB | $2 *$ NSTP | $\begin{aligned} & \text { 64+EXTRA+ } \\ & 2 * \text { NDS }+ \end{aligned}$ | SPH element block start and end numbers |


| VALUE | \#WORDS | DISK | DESCRIPTION |
| :--- | :--- | :--- | :--- |
|  |  | ADDRESS |  |
|  |  | $2 *$ NSTH + |  |
|  |  | $2 *$ NSTB+ |  |
|  |  | $2 *$ NSTS+ |  |
|  |  | $2 *$ NSTT |  |

ISTRN can only be computed as follows and if NSTS $>0$.
If NV2D-MAXINT*(6*IOSHL(1)+IOSHL(2)+NEIPS)+8*IOSHL(3)+4*IOSHL(4) > 10
Then ISTRN $=1$, else ISTRN $=0$
Or NSTT > 0
If NV3DT-MAXINT*( $6 * \operatorname{IOSHL}(1)+\operatorname{IOSHL}(2)+$ NEIPS $)>10$
Then ISTRN $=1$, else ISTRN $=0$

## SMOOTH PARTICLE HYDRODYNAMICS ELEMENT DATA FLAGS

This section is only output if NMSPH $>0$. The section is a list of flags to indicate what SPH data is output for each SPH node/element. The first number is the length in words for this array, currently $=10$.
SPH elements are centered at nodes, and cover a spherical volume defined by the radius of influence. They do not have a connectivity with other SPH elements. They should be displayed as a dot or a spherical surface, with radius scaling to reduce the size and enable each element to be distinguishable.
As follows:

$$
\begin{aligned}
& \operatorname{isphfg}(1)=10-\text { length of sph flags array } \\
& \text { isphfg }(2)=1 \quad \text { - radius of influence } \\
& \text { isphfg }(3)=1 \quad \text { - pressure in particle } \\
& \text { isphfg(4)=6 }
\end{aligned}-6 \text { true stress components } \quad \begin{array}{ll}
\text { isphfg }(5)=1 & \text { - plastic strain, }>0.0 \text { if effective stress exceeds yield strength } \\
\text { isphfg}(6)=1 & \text { - density of particle material } \\
\text { isphfg }(7)=1 & \text { - internal energy (strain) } \\
\text { isphfg }(8)=1 & \text { - number of neighbors affecting particle } \\
\text { isphfg }(9)=6 & -6 \text { true strain components } \\
\text { isphfg }(10)=1 & \text { - mass of element }
\end{array}
$$

If the value of isphfg $(2-10)=0$, then the particular data item is not output for the particle. To calculated the size of data add the isphfg values from isphfg(2) through isphfg(10) and add one. One value is always output which is the material number as a floating point number for each particle.
If this value is negative then the particle has been deleted from the model.
Full output for each particle is:
 Hence total size is 20 .
When a particle is deleted from the model, data is still output for it because the length of data must always be the same for each state.

## GEOMETRY DATA

The geometry section contains the nodal coordinates and the element connectivities. The ordering of the nodal points is assumed to be the same as the ordering of the nodal data in the state data that follows. The connectivities are assumed to be packed with 3 integers per word unless NDIM is set to 4 as in the new LS-DYNA/3D, LS-NIKE3D databases. The order of the elements are 3,2 , and 1 dimensional elements if the database is $\mathrm{ICODE}=2$ or 6 .

VALUE
$\mathrm{X}(3,1)$
IX8(9,1)
$\operatorname{IXT}(9,1)$

IX2 $(6,1)$

IX4(5,1)

LENGTH
NDIM*NUMNP

9*NEL8

9*NELT

6*NEL2

5*NEL4

## DESCRIPTION

Array of nodal coordinates $\mathrm{X} 1, \mathrm{Y} 1, \mathrm{Z1}$, X2,Y2,Z2,X3,Y3,Z3, ... ,Xn,Yn,Zn
Connectivity and material number for each 8 node solid element

Connectivity and material number for each 8 node thick shell element
Connectivity, orientation node, two null entries, and the material number for each 2 node beam element

Connectivity and material number for each 4 node shell element

## USER MATERIAL, NODE, AND ELEMENT IDENTIFICATION NUMBERS

Skip this section if NARBS (disk address 39) is zero. The user node and element numbers must be in ascending order. It is assume that if this option is used all the node and element data in the databases is in ascending order in relation to the user numbering. For sequential material/part numbering, the total length of data is:

NARBS $=10+$ NUMNP+NEL8+NEL2+NEL4+NELT +
3*NMMAT : these numbers are not used

For arbitrary material numbering (NSORT < 0)
NARBS $=16+$ NUMNP+NEL8+NEL2+NEL4+NELT+3*NMMAT
Where material numbers are not in ascending order.

| VALUE | LENGTH | DESCRIPTION |
| :--- | :--- | :--- |
| NSORT | 1 | Pointer to arbitrary node numbers in <br> LS-DYNA/3D code, If $<0$, it flags that arbitrary <br> material identification numbers are also used. |
| NSRH | 1 | Pointer to arbitrary solid element numbers in <br> LS-DYNA code: <br> =NSORT+NUMNP |
| NSRB | 1 | Pointer to arbitrary beam element numbers in <br> LS-DYNA code: <br> $=$ NSRH+NEL8 |
|  | 1 | Pointer to arbitrary shell element numbers in <br> LS-DYNA code: <br> =NSRB+NEL2 |
| NSRS | 1 | Pointer to arbitrary thick shell element numbers in <br> LS-DYNA code: |
|  | 1 | =NSRS+NEL4 <br> NSRT |
|  | 1 | Number of nodal points <br> Number of 8 node solid elements |
| NSRHD | 1 | Number of 2 node beam elements |
| NSRBD | 1 | Number of 4 node shell elements |
| NSRSD | 1 | Number of 8 node thick shell elements |
| NSRTD |  |  |


| NSRMA | 1 | Pointer to an array in the LS-DYNA code that list <br> the material ID's in ascending order. |
| :--- | :--- | :--- |
| NSRMU | 1 | Pointer to an array in the LS-DYNA code that gives <br> the material ID's in the actual order that they are <br> defined in the user input. |
| NSRMP | 1 | Pointer to an array in the LS-DYNA code that gives <br> the location of a member in the array originating at <br> NSRMU for each member in the array starting at |
| NSRMA. |  |  |

## TIME HISTORY DATA

The time database contains the following data:

- Time word
- Node data
- Node data for solids, thick shells, and shells, respectively
- Element data for solids, thick shells, beams, and shells, respectively

| VALUE | LENGTH | DESCRIPTION |
| :---: | :---: | :---: |
| TIME | 1 | Time word |
| GLOBAL | NGLBV | Global variables for this state. |
|  |  | LS-DYNA Global Variables: |
|  |  | KE, IE, TE, X, Y, and Z velocity |
|  |  | IE for each part |
|  |  | KE for each part |
|  |  | $\mathrm{X}, \mathrm{Y}$, and Z velocity for part 1 |
|  |  | $\ldots$ |
|  |  | $\mathrm{X}, \mathrm{Y}$, and Z velocity for part n |
|  |  | Mass for each part |
|  |  | Hourglass energy for each part |
|  |  | Force for each rigid wall |
|  |  | $=6+7 *$ (NUMMAT8+ NUMMAT2 + NUMMAT4 |
|  |  | + NUMMATT + NUMRBS) + N*NUMRW, N=1 or |
|  |  | $\mathrm{N}=4$ (ls971) |

SKIP THE FOLLOWING DATA IF THE NUMBER OF NODE BLOCKS FOR PLOTTING IS ZERO (VALUE NUMDS AT DISK ADDRESS 25)

| TIME | 1 | Time word |
| :--- | :--- | :--- |
| NODEDATA | NND | Total nodal values for state where <br> NLN $=10 *$ TNODS where TNODS is the number <br> of nodes put into database. The database contains <br> TNODS vectors each with up to 10 components: <br> temperature (if $I T=1) ; x, y$, and $z$ coordinates; $x$, <br> $y$, and $z$ velocities; and $x, y$, and $z$ accelerations. |

CFDDATA
CFD
Bit flag: NCFDV1, bits from right to left Eg Pressure, Resultant Vorticity, and Density NCFDV1 $=2+32+1024=1058$

14 Pressure
15 X Vorticity
16 Y Vorticity
17 Z Vorticity
18 Resultant Vorticity
19 Enstrophy
20 Helicity
21 Stream Function
22 Enthalpy
23 Density
24 Turbulent KE
25 Dissipation
14-20 Eddy Viscosity
Bit flag: NCFDV2
2-11 Species 1 through 10
Count number of bits on * NUMNP

SKIP THE FOLLOWING DATA IF THE NUMBER OF ELEMENT BLOCKS FOR IS ZERO (VALUE NUMDS AT DISK ADDRESS 26)
****SKIP THE FOLLOWING IF THERE IS NO DATA FOR SOLID ELEMENTS

TIME

SOLIDDATA

1

ENV

Time word

Total nodal values for solid elements where ENV=56*TBELM where THELM is the total number of solid elements to be put into the database. The data contains THELM vectors each with 56 components ordered as follows: 8 connectivities: $\mathrm{x}, \mathrm{y}, \mathrm{z}$ coordinates for each of the 8 nodes; and, lastly, $\mathrm{x}, \mathrm{y}, \mathrm{z}$ velocities for each of the 8 nodes.

For solid elements the database contains (7+NEIPH-6*ISTRN) values per element. One set of global stresses are always put into the database for each solid element followed by NEIPH history values. Only data for elements defined in the time history blocks is output. The ordering of the data follows:

1. Sigma-x (true stress in the global system)
2. Sigma-y
3. Sigma-z
4. Sigma-xy
5. Sigma-yz
6. Sigma-zx
7. Effective plastic strain or material dependent variable
8. First extra value (if NEIPH $>0$ )
9. Second extra value (if NEIPH $>1$ )
10. Etc. until NEIPH extra values are defined if ISTRN=1

7+NEIPH-5. Epsilon-x
7+NEIPH-4. Epsilon-y
7+NEIPH-3. Epsilon-z
7+NEIPH-2. Epsilon-xy
7+NEIPH-1. Epsilon-yz
7+NEIPH. Epsilon-zx

## ****SKIP THE FOLLOWING IF THERE IS NO DATA FOR THICK SHELLS ****

TIME

TSHELLDATA
ENV

Time word

Total nodal values for thick shell elements where ENV=56*TBSEL where TBSEL is the total number of thick shell elements in the database. The data contains TBSEL vectors each with 56 components ordered as follows: 8 connectivities: $\mathrm{x}, \mathrm{y}, \mathrm{z}$ coordinates for each of the 8 nodes; and, lastly, $x, y, z$ velocities for each of the 8 nodes.

## ****SKIP THE FOLLOWING IF THERE IS NO DATA FOR SHELL ELEMENTS ****

TIME

SHELLDATA

1

ENVS

Time word

Total nodal values for shell elements where ENVS $=28 *$ TSELM where TSELM is the total number of shell elements in the database. The data contains TSELM vectors each with 28 components ordered as follows: 4 connectivities: $\mathrm{x}, \mathrm{y}, \mathrm{z}$ coordinates for each of the 4 nodes; and, lastly, $\mathrm{x}, \mathrm{y}, \mathrm{z}$ velocities for each of the 4 nodes.

## ****SKIP THE FOLLOWING IF THERE IS NO DATA FOR THICK SHELLS ****

For thick shell elements the database contains NV3DT values per element. Three sets of global stresses are always put into the database for each thick shell and are located at the mid surface, the inner surface, and the outer surface, respectively. If one integration point is used the single state is written three times. If two integration points are used then the mid surface value is taken as the average value. The inner values of the stress are always set to the values at the innermost integration point and likewise for outer values. If no integration point lies at the center, i. e. an even number of integration points through the thickness, a value is computed that is an average of the two integration point lying nearest the mid surface. Only data for elements defined in the time history blocks is output. The ordering of the data follows:

1. Sigma-x (mid surface true stress in global system)
2. Sigma-y
3. Sigma-z
4. Sigma-xy
5. Sigma-yz
6. Sigma-zx
7. Effective plastic strain or material dependent variable
*. Define NEIPS additional history values here for midsurface
8. Sigma-x (inner surface true stress in global system)
9. Sigma-y
10. Sigma-z
11. Sigma-xy
12. Sigma-yz
13. Sigma-zx
14. Effective plastic strain or material dependent variable
*. Define NEIPS additional history values here for inner surface
15. Sigma-x (outer surface true stress in global system)
16. Sigma-y
17. Sigma-z
18. Sigma-xy
19. Sigma-yz
20. Sigma-zx
21. Effective plastic strain or material dependent variable
*. Define NEIPS additional history values here for outer surface
22. Effective plastic strain or material dependent variable
*. Define NEIPS additional history values here for outer surface
If MAXINT $>3$ then define an additional (MAXINT-3 ) ( $6 * \operatorname{IOSHL}(1)+1 * \operatorname{IOSHL}(2)+$ NEIPS $)$ quantities here
*. If ISTRN=1, then define strain components Epsilon ( $\mathbf{x}, \mathbf{y}, \mathbf{z}, \mathbf{x y}, \mathbf{y z}, \mathbf{z x}$ ) here for inner surface and outer surface

## SKIP THE FOLLOWING IF THERE IS NO DATA FOR BEAM ELEMENTS

TIME

BEAMDATA

1

BESV

Time word

Total element values for beam elements where BESV=NV1D*TBELM. The data contains TBELM vectors each with NV2D values ordered as described below.

For beam elements the database contains NV1D $=6$ values per element. They are:

1. Axial force
2. S shear resultant
3. T shear resultant
4. $\quad \mathrm{S}$ bending moment
5. T bending moment
6. Torsional resultant

If there are values output at beam integration points, then NV1D $=6+5 *$ BEAMIP
7. RS shear stress
8. TR shear stress
9. Axial stress
10. Plastic strain
11. Axial strain

BEAMIP is set in *DATABASE_EXTENT_BINARY
****SKIP THE FOLLOWING IF THERE IS NO DATA FOR SHELL ELEMENTS ****
For shell elements the database contains NV2D values per element. If the minimum value of MAXINT is 3 , then the stresses are typically located at the mid surface, the inner surface, and the outer surface, respectively. If one integration point is used the stress is written three times. If two integration points are used then the mid surface value is taken as the average value. The inner values of the stress are always set to the values at the innermost integration point and likewise for outer values. If no integration point lies at the center, i. e. an even number of integration points through the thickness, a value is computed that is an average of the two integration point lying nearest the mid surface. Only data for elements defined in the time history blocks is output. The ordering of the data follows:

1. Sigma-x (mid surface true stress in global system)
2. Sigma-y
3. Sigma-z
4. Sigma-xy
5. Sigma-yz
6. Sigma-zx
7. Effective plastic strain or material dependent variable
*. Define NEIPS additional history values here for midsurface
8. Sigma-x (inner surface true stress in global system)
9. Sigma-y
10. Sigma-z
11. Sigma-xy
12. Sigma-yz
13. Sigma-zx
14. Effective plastic strain or material dependent variable
*. Define NEIPS additional history values here for inner surface
15. Sigma-x (outer surface true stress in global system)
16. Sigma-y
17. Sigma-z
18. Sigma-xy
19. Sigma-yz
20. Sigma-zx
21. Effective plastic strain or material dependent variable
*. Define NEIPS additional history values here for outer surface
If MAXINT $>3$ then define an additional (MAXINT-3 ) ${ }^{*}$ ( 6 $\operatorname{IOSHL}(1)+1 * \operatorname{IOSHL}(2)+$ $8 * \operatorname{IOSHL}(3)+4 * \operatorname{IOSHL}(4)+\mathrm{NEIPS})$ quantities here
22. Bending moment-mx (local shell coordinate system)
23. Bending moment-my
24. Bending moment-mxy
25. Shear resultant-qx
26. Shear resultant-qy
27. Normal resultant-nx
28. Normal resultant-ny
29. Normal resultant-nxy
30. Thickness
31. Element dependent variable
32. Element dependent variable
33. Internal energy (if and only if ISTRN=0)

The following quantities are expected if and only if ISTRN=1
33. eps-x (inner surface strain in global system)
34. eps-y
35. eps-z
36. eps-xy
37. eps-yz
38. eps-zx
39. eps-x (outer surface strain in global system)
40. eps-y
41. eps-z
42. eps-xy
43. eps-yz
44. eps-zx
45. Internal energy (if and only if ISTRN=1)
****SKIP THE FOLLOWING IF THERE IS NO DATA FOR SPH ELEMENTS ****

| TIME | 1 |
| :--- | :--- |
| SPHDATA | SPHV |

Time word

Data for each sph element according to the sph flags SPHV=NSTP*NUM_SPH_DATA

## INTERFACE FORCE DATABASE

There are three sections in the interface force database. The first contains 64 words of control information. The second contains geometric information, i.e. the nodal coordinates and segment connectivities for each segment contained in the master and slave surface definitions. The third section contains the results of the analysis at sequential output intervals. The output at a given time is called a state. The state contains a time word, global variables such as total energies and momenta, nodal data consisting of accelerations, velocities, and displacements, and finally segment data is written that include the pressure and shear stress acting on each segment and nodal forces for each node that defines the segment. The control information that follows provides information as to what is in the file and which database is being processed.

## CONTROL DATA

## DISK

VALUE \#WORDS ADDRESS DESCRIPTION

| Title | 10 | 0 | Problem identification |
| :---: | :---: | :---: | :---: |
| Run time | 1 | 10 | time in seconds since 00:00:00 UTC, January 1, 1970 |
| File type | 1 | 11 | intfor $=4$ <br> $1=\mathrm{d} 3$ plot, $2=\mathrm{d} 3 \mathrm{drlf}$, $3=\mathrm{d} 3$ thdt, $4=$ intfor, $5=\mathrm{d} 3$ part <br> $6=$ blstfor, $7=\mathrm{d} 3 \mathrm{cpm}, 8=\mathrm{d} 3 \mathrm{ale}, 11=\mathrm{d} 3 \mathrm{eigv}$, <br> $12=\mathrm{d} 3$ mode, $13=\mathrm{d} 3$ iter, $21=\mathrm{d} 3 \mathrm{ssd}, 22=\mathrm{d} 3 \mathrm{spcm}$, <br> $23=\mathrm{d} 3 \mathrm{psd}, 24=\mathrm{d} 3 \mathrm{rms}, 25=\mathrm{d} 3 \mathrm{ftg}, 26=\mathrm{d} 3 \mathrm{acs}$ |
| Source version | 1 | 12 | ls-dyna version *1000000 + svn number |
| Release number | 1 | 13 | Release number in character*4 form 50 for R5.0 <br> 511c for R5.1.1c |
| Version | 1 | 14 | Code version |
| NDIM | 1 | 15 | Insert 4 for LS-DYNA/3D database |
| NUMNP | 1 | 16 | Number of nodal points |
| ICODE | 1 | 17 | Insert 6 for LS-DYNA/3D database |
| NGLBV | 1 | 18 | Number of global variable to be read |
| BLANK | 1 | 19 | Insert zero |
| IU | 1 | 20 | Flag for current geometry (=1) |


| IV | 1 | 21 | Flag for velocities (default=1) |
| :---: | :---: | :---: | :---: |
| BLANK | 1 | 22 | Insert zero |
| BLANK | 1 | 23 | Insert zero |
| BLANK | 1 | 24 | Insert zero |
| BLANK | 1 | 25 | Insert zero |
| BLANK | 1 | 26 | Insert zero |
| BLANK | 1 | 27 | Insert zero |
| BLANK | 1 | 28 | Insert zero |
| BLANK | 1 | 29 | Insert zero |
| BLANK | 1 | 30 | Insert zero |
| NUMSG | 1 | 31 | Total number of slave and master segments in sliding interface definitions. |
| NUMMAT4 | 1 | 32 | $=2$ times the number of sliding interfaces. |
| NV2D | 1 | 33 | $=$ Number of values for each 2D segment. <br> If NV2D is negative then the file is FSIFOR for an ALE model |
| BLANK | 5 | 34 | Unused space |
| NARBS | 1 | 39 | Additional storage required for arbitrary node and element numbering in type 6 database This number equals the sum of (10+ NUMNP+NEL8+NEL2+NEL4+ NELT) |
| BLANK | 19 | 40 | Unused space |
| NWEAR | 1 | 59 | 0, 4 or 8 |
| NPRESU | 1 | 60 | $0,1,2$, or 3 |
| NSHEAR | 1 | 61 | 0 or 3 |
| NFORCE | 1 | 62 | 0 or 12 |
| NGAPC | 1 | 63 | 0 or 5 |

If *DATABASE_EXTENT_INTFOR is included in the model input the following values apply to the state output:
NV2D $=$ NWEAR + NPRESU + NSHEAR + NFORCE + NGAPC
NWEAR: output surface wear at each node of segments
EQ. 0 no values
EQ. 4 output wear depth for each node

EQ. 8 output wear depth and sliding distance for each node
NPRESU: output option for pressures
EQ. 0 no pressures output
EQ. 1 output normal interface pressure only
EQ. 2 output normal interface pressure and peak pressure
EQ. 3 output normal interface pressure, peak pressure and time to peak pressure

NSHEAR: output option for maximum interface shear stress, shear stress in r-direction and s-direction
EQ. 0 no
EQ. 3 yes

NFORCE: output option for X-, Y- and Z-force at all 4 nodes
EQ. 0 no
EQ. 12 yes

NGAPC: output option for contact gap at all nodes and surface energy density EQ. 0 no
EQ. 5 yes

## GEOMETRY DATA

The geometry section contains the nodal coordinates and the element connectivities. The ordering of the nodal points is assumed to be the same as the ordering of the nodal data in the state data that follows.

VALUE
$\mathrm{X}(3,1)$

IX4 $(5,1)$

LENGTH
NDIM*NUMNP

5*NUMSG

## DESCRIPTION

Array of nodal coordinates $\mathrm{X} 1, \mathrm{Y} 1, \mathrm{Z1}$, X2,Y2,Z2,X3,Y3,Z3, ... ,Xn,Yn,Zn
Connectivity and identification number for each 3 or 4 node interface segment. For sliding interface $n$ the identification number in $2 \mathrm{n}-1$ for the slave surface and 2 n for the master surface.

## USER MATERIAL, NODE, AND ELEMENT IDENTIFICATION NUMBERS

Skip this section if NARBS (disk address 39) is zero. The user node and element numbers must be in ascending order. It is assumed that if this option is used all node and element data anywhere in the databases is in ascending order based on user numbering. For sequential material/part numbering, the total length of data is:

NARBS $=10+$ NUMNP+NEL8+NEL2+NEL4+NELT +
3*NMMAT : these numbers are not used

For arbitrary material numbering (NSORT < 0)
NARBS $=16+$ NUMNP+NEL8+NEL2+NEL4+NELT+3*NMMAT
Where material numbers are not in ascending order.

| VALUE | LENGTH | DESCRIPTION |
| :--- | :--- | :--- |
| NSORT | 1 | Pointer to arbitrary node numbers in <br> LS-DYNA code, If $<0$, it flags that arbitrary <br> material identification numbers are also used. |
| NSRH | 1 | Pointer to arbitrary solid element numbers in <br> LS-DYNA code: <br> =NSORT+NUMNP |
| NSRB | 1 | Pointer to arbitrary beam element numbers in <br> LS-DYNA code: <br> =NSRH+NEL8 |
| NSRS | 1 | Pointer to arbitrary shell element numbers in <br> LS-DYNA code: <br> =NSRB+NEL2 |
| NSRT | 1 | Pointer to arbitrary thick shell element numbers in <br> LS-DYNA code: <br> =NSRS+NEL4 |
| NSORTD | 1 | Number of nodal points <br> NSRHD |
| NSRBD | 1 | Number of 8 node solid elements <br> Number of 2 node beam elements <br> Number of 4 node shell elements |
| NSRSD | 1 | 1 |

\(\left.$$
\begin{array}{lll}\text { NSRMA } & 1 & \begin{array}{l}\text { Pointer to an array in the LS-DYNA code that list } \\
\text { the contact ID's in ascending order. }\end{array} \\
\text { NSRMU } & 1 & \begin{array}{l}\text { Pointer to an array in the LS-DYNA code that gives } \\
\text { the contact ID's in the actual order that they are } \\
\text { defined in the user input. }\end{array} \\
\text { NSRMP } & 1 & \begin{array}{l}\text { Pointer to an array in the LS-DYNA code that gives } \\
\text { the location of a member in the array originating at }\end{array}
$$ <br>

NSRMU for each member in the array starting at\end{array}\right\}\)| NSRMA. |
| :--- |
| NSRTM |

## ALE, CPM and DES Output Fringe Labels

At the end of the first binary files, eg intfor, fringe labels are appended.
This extra data is written at the end of the following files:
ALE, CPM and DES intfor files. Labels for fringe titles are written directly after the EOF (= -999999.0) marker.

Header output

| ------------------------------- |  |  |
| :--- | :--- | :--- |
| NTYPE | 1 | entity type $=90200$ |
| NLABEL | 1 | number of labels |
| NCHAR8 | 1 | number of char* 8 characters in each label |
| LABEL | NCHAR8 | Fringe label |

For the interface force file (intfor), Fringe labels are written at the end of first file after the EOF (= -999999.0) marker

## STATE DATA

The state data for the interface forces have three parts:

- Time word and global data
- Node data
- Force data for sliding interface segments

| VALUE | LENGTH | DESCRIPTION |
| :--- | :--- | :--- |
| TIME | 1 | Time word |
| GLOBAL | NGLBV | Global variables for this state |
| NODEDATA | NND | Total nodal values for state where <br> NND=(IT+NDIM*(IU+IV))*NUMNP LS- <br> DYNA/3D writes 6 values per node, i.e., the <br> three coordinates and the translational <br> velocities. The Data is put into the database <br> as two vectors: first X(3,NUMNP) and then |
|  |  | V(3,NUMNP), respectively. <br> Data for sliding interface segments <br> where the quantity |
|  |  | ENN = NUMSG*NV2D. <br> SEGMDATA |
|  | ENN organization of the segment data is |  |
| described below. |  |  |

This state data is repeated for each state in the database.

For each sliding interface segment the database contains NV2D values per segment. The data order is:

1. Normal interface pressure acting on segment
2. Maximum interface shear stress acting on segment
3. Shear stress in local r-direction of segment
4. Shear stress in local s-direction of segment
5. X force at node n 1 of segment
6. Y force at node n 1 of segment
7. Z force at node n 1 of segment
8. $\quad \mathrm{X}$ force at node n 2 of segment
9. Y force at node n 2 of segment
10. Z force at node n 2 of segment
11. X force at node n 3 of segment
12. Y force at node n 3 of segment
13. Z force at node n 3 of segment
14. X force at node n 4 of segment
15. Y force at node $n 4$ of segment
16. Z force at node n 4 of segment
17. contact gap at node n 1
18. contact gap at node n 2
19. contact gap at node n 3
20. contact gap at node $n 4$
21. surface energy density of segment
22. peak pressure
23. time to peak pressure
24. surface wear depth at node n 1
25. surface wear depth at node n 2
26. surface wear depth at node n3
27. surface wear depth at node $n 4$
28. wear sliding distance at node n 1
29. wear sliding distance at node n 2
30. wear sliding distance at node n3
31. wear sliding distance at node n 4

If contact gap at node $n 1=-1.0$, then no values are set. Similarly, if surface energy density is $=$ -1.0 , no value is set.

If contact gap is $>=999.0$ ignore the value, this means that the interfaces are not in discernible contact.

Note: original interface force files have 16 variables, while recent ones, from 1s-dyna 1s970 and ls971, have up to 31. Check size of NV2D. If *database_extent_intfor is include in the model input then NV2D is according to the formula above.

## FSIFOR file output:

1. Normal interface pressure acting on segment
2. X force on segment
3. Y force on segment
4. $\quad \mathrm{Z}$ force on segment
5. relative interface velocity
6. X interface velocity
7. Y interface velocity
8. Z interface velocity

## BLSTFOR file (NV2D=16 or 7) output:

1. effective (combined incident and reflected) pressure applied to the segment
2. reflected wave (relevant only for BLAST=4)
$=-1$ : segment is below ground level and not exposed to blast
$=0$ : segment has not been subjected to blast waves
$=1:$ segment has been subjected to the initial incident wave
$=2$ : segment has been subjected to the ground reflected wave
$=3$ : segment resides in the Mach stem region (Since known by geometrical
considerations this value is fixed at time $t=0$. Thus, any non-zero reflected pressure on this segment is due to the Mach wave.)
3. incident pressure
4. mass density of air
5. global $x$-velocity of blast wind
6. global y-velocity of blast wind
7. global z-velocity of blast wind

## CRACK FILE (d3crck)

The crack file is created in LS-DYNA when the Winfrith Concrete material model is used for solid elements. This model allows up to three orthogonal crack planes to develop each with an origin at the center of the element. The plane is assumed to project to the surface of the solid, and can be represented by drawing a line on any of the six solid element faces where it emerges. This cutting line can be found by considering the intersection of each plane and each solid face.

The crack file is written as a Fortran unformatted binary file, and each record in the file has a start record mark and an end record mark, each of which is 4 bytes for a single precision run and 8 bytes for a double precision run. The data is output for a state and contains:

| VALUE | \#WORDS | DESCRIPTION |
| :--- | :--- | :--- |
|  |  |  |
| TIME | 1 | State Time word |
| NUMBER OF CRACKS | 1 | NC, Number of sets of crack data |
| CRACK DATA | $16 *$ NC | Sets of data for each crack |

Data for each crack contains: Element ID, Flag for each crack plane, Normal vector for each crack plane and width for each crack plane.
Crack plane flags are as follows:
$0=$ no crack, $1=$ cracked, but no sustaining tensile load, $2=$ cracked but closed up, and $3=$ fully cracked with no tensile strength.
Shown below is some C coding from LS-PREPOST to illustrate how the crack data is used and the crack lines established. The important coding is picked out in bold type. Also, after the crack data was read in, the crack flags (3 words) where stored as bits in one word. NG and MAT are the Group number for elements and the Material ID we assign in LS-PREPOST.

```
/* elm id, 3 2bit flags 0-> 2, group no., mat no. */
int id, pflag, ng, mat;
float abc[9]; /* plane normals */
float cp[3]; /* crack width planes 1,2 and 3 */
```

```
/* check brick element face against crack plane ( \(\mathrm{nx}, \mathrm{ny}, \mathrm{nz}\) ) positioned
    * at element center (xe,ye,ze)
*/
int CrackPlane(float xe, float ye, float ze, float nx, float ny, float nz,
    float \(\mathrm{xyz}[\mid[3]\), float \(\mathrm{xs}[2]\), float \(\mathrm{ys}[2]\), float \(\mathrm{zs}[2]\) )
```

```
{
    int i, k;
    float x0, y0, z0, x1, y1, z1;
    float xp1, yp1, zp1, xp2, yp2, zp2, pn, dn, t;
    k=0;
    x0 = xyz[3][0];
    y0 = xyz[3][1];
    z0 = xyz[3][2];
    for (i=0; i<4; i++) {
        /* parametric line clip algorithm */
        xp1 = x0 - xe;
        yp1 = y0 - ye;
        zp1 = z0-ze;
        x1 = xyz[i][0];
        y1 = xyz[i][1];
        z1 = xyz[i][2];
        xp2 = x1 - x0;
        yp2 = y1 - y0;
        zp2 = z1 - z0;
        x0 = x1;
        y0 = yl;
        z0 = z1;
        pn = nx*xp1 + ny*yp1 + nz*zp1;
        dn = nx*xp2 + ny*yp2 + nz*zp2;
        if (fabs(dn) < 1.0e-15) continue;
        t = -pn / dn;
        if (t < 0.0 || t> 1.0) continue;
        t = t - 1.0;
        xs[k] = x1 + t * xp2;
        ys[k]= yl +t * yp2;
        zs[k] = z1 + t * zp2;
        k++;
        if (k>1) break;
    }
    return k;
}
void SetCrackWidth(float v)
{
    min_crack_width = MAX(0.0, v);
}
void DrawCracks(int ist, float *bg_color)
{
    int i, k, m, n, nc, nd, kd, ip, ic, is;
    int id, facecode;
    int etype, nface, pflag, flag;
    unsigned int j;
    float xc, yc, zc, a, b, c;
    float xyz[24][3];
    float xi[2], yi[2], zi[2];
    int shrink, count, ns[2];
    float dx, dy, dz, ds, d;
```

```
NDCOOR *nod;
int ng=0;
float rd, gn, bu;
nod = node;
nod--;
rd = 1.0 - bg_color[0];
gn = 1.0 - bg_color[1];
bu = 1.0 - bg_color[2];
glDisable(GL_LIGHTING);
glColor3f(rd, gn, bu);
glLineWidth(2.0);
GetCrackData(ist);
glBegin(GL_LINES);
nc = cstate[ist].nc;
for (n=0; n<nc; n++) {
    pflag = crack[n].pflag;
if (pflag == 0) continue;
id = crack[n].id - 1;
k = solid[id].mat & MASKBIT22;
j = active_list[k].loc;
facecode = (active_list[k].akey>>2) & FACEBITS1;
if (j < BIT30 && facecode > 0) {
    id = j;
    ng = crack[n].ng;
            if (!part[ng].active) continue;
    dx = part[ng].dscale[0];
    dy = part[ng].dscale[1];
    dz = part[ng].dscale[2];
    ds = part[ng].dscale[3];
    shrink = part[ng].shrink_mode;
    etype = (active_list[k].akey) & 0x3;
    nface = FACE_NF[etype];
    xc = yc = zc = 0.0; ic = 0;
    for (i=0; i<nface; i++) {
        for (m=0; m<4; m++) {
            nd = FACE_P[etype][i][m];
            kd = solid[id].conn[nd];
            k=4 * i + m;
            if (ds == 0.0) {
                    xyz[k][0] = (disp_state+kd)->xyz[0];
                    xyz[k][1] = (disp_state+kd)->xyz[1];
                    xyz[k][2] = (disp_state+kd)->xyz[2];
            }
            else {
                    GetScaledNodalCoord(kd, nod, disp_state, dx,dy,dz, xyz[k]);
            }
            xc += xyz[k][0];
            yc += xyz[k][1];
            zc += xyz[k][2];
                ic += 1;
```

```
        }
            }
            if (ic == 0) continue;
            d = 1.0 / (float)ic; xc *= d; yc *= d; zc *= d;
            if (shrink) {
                for (i=0; i<24; i++) {
                xyz[i][0] = xc + (xyz[i][0]- xc) * shrink_factor;
                xyz[i][1] = yc + (xyz[i][1] - yc) * shrink_factor;
                    xyz[i][2] = zc + (xyz[i][2]-zc)* shrink_factor;
                }
            }
        for (i=0; i<nface; i++) {
        if (shrink || (facecode & FACE_CODE1[i])) {
            k=4 * i;
            /* for each active crack plane */
            for (ip=0,is=0,ic=0; ip<3; ip++,is+=2,ic+=3) {
                flag = (pflag>>is) & 3;
                if (flag == 0) continue;
                if ((min_crack_width < 0.5 && crack[n].cp[ip] >= min_crack_width)
                    |(min_crack_width >= 0.5 && flag == 3)) {
                    a = crack[n].abc[ic];
                    b = \operatorname{crack[n].abc[ic+1];}
                    c = crack[n].abc[ic+2];
                    count = CrackPlane(xc, yc, zc, a, b, c, &xyz[k], xi, yi, zi);
                    if (count > 1) {
                        glVertex3f(xi[0], yi[0], zi[0]);
                    glVertex3f(xi[1], yi[1], zi[1]);
                    }
                }
            }
        }
        }
        }
    }
    gIEnd();
    glLineWidth(1.0);
}
```


## DYNAIN BINARY FILE FORMAT (dynain.bin)

/* Discription of Dynain binary format:

* In first 100 words (integers)
* head[0] = location of nodal data
* head[1] = number of nodes
* head[2] = location of solid element connectivities
* head[3] = number of solid elements
* head[4] = location of shell element connectivities + thicknesses
* head[5] = number of shell elements
* head[6] = location of adaptive constraints
* head[7] = number of adaptive constraints
* head[8] = location of initial stresses for solid elements
* head[9] = number of initial stress states defined for solids
* head[10] = location of initial stresses for shell elements
* head[11] = number of initial stress states defined for shells
* head[12] = location of initial strains for shell elements
* head[13] = number of initial strains states defined for shells
* head[14] = location of boundar spc's
* head[15] = number of boundary spc's
* head[16] = location of local coordinate systems by nodes
* head[17] = number of local coordinate systems by nodes
* head[18] = location of local coordinate systems by vector
* head[19] = number of local coordinate systems by vector
* head[20] = location of initial stress states for beams
* head[21] = number of initial stress states for beams
* head[22] = location of thick shell element connectivities
* head[23] = number of thick shell elements
* head[24] = location of initial stresses for thick shell elements
* head[25] = number of initial stress states defined for thick shells
* head[26] = location of beam element connectivities
* head[27] = number of beam elements
* head[28] = location of initial strains for solid elements
* head[29] = number of initial strain states defined for solids
*/


# EXTRA DATA TYPE DEFINITIONS (NCFDV1 = 67108864) 

```
#ifndef HAVE D3PLOT
#define '-HAVE-D3PLOT 1
#define D3PL_FIRST SCALAR ID 0
#define D3PL-FIRST }\mp@subsup{}{}{-}\mathrm{ VECTOR ID 1000
#define D3PL_FIRST_TENSOR_ID 2000
#define D3PL_END_IDS 3000
/* scalar variable names */
enum {
    D3PL Pressure INS=0,
    D3PL_Temperature_INS,
    D3PL_Enstrophy_INS,
    D3PL-Helicity INS,
    D3PL_Stream_function_INS,
    D3PL_Enthalpy_INS,
    D3PL Turbulent KE INS,
    D3PL_Turbulent_eps_INS,
    D3PL_Eddy_Viscosity_INS,
    D3PL Density_INS,
    D3PL-VolFracÉSpec1 INS,
    D3PL_VolFractSpec2_INS,
    D3PL-VolFractSpec3-INS,
    D3PL-VolFractSpec4-INS,
    D3PL_VolFractSpec5_INS,
    D3PL VolFractSpec6 INS,
    D3PL-}VolFractSpec7-INS
    D3PL_VolFractSpec8_INS,
    D3PL VolFractSpec9 INS,
    D3PL_VolFractSpec1\overline{0}_INS,
    D3PL_Density_CESE,
    D3PL Pressure CESE,
    D3PL-Temperatūre CESE,
    D3PL-Total energy CESE,
    D3PL_Internal_energy_CESE,
    D3PL Enthalpy CESE,
    D3PL_Entropy_\overline{CESE,}
    D3PL_Stream_function_CESE,
    D3PL Density TS CESE,
    D3PL-Total eñerg}y TS CESE
    D3PL_Temperature_radflow,
    D3PL_Intensity radflow,
    D3PL-Scalar potential,
    D3PL_Electrical_conductivity,
    D3PL_Ohm_heating_power_FEM,
    D3PL-
    D3PL_Temperature-PFEM,
    D3PL_Pressure_PFEM,
    D3PL K PFEM,
    D3PL_eps PFEM,
    D3PL_particle_size,
    D3PL_particle_temperature,
    D3PL_particle_cnt_child_particles,
    D3PL_Vorticity_PFEM,
    D3PL Cp PFEM,
    D3PL_QC-PFEM,
    D3PL_Shear_PFEM,
```

```
    D3PL void fraction CESE,
    D3PL_Schlieren_number_CESE,
    D3PL_LEVELSET_PEEM,
    D3PL-AVG PRES PFEM,
    D3PL_
    D3PL_relative_mu,
    D3PL-HCC PFEM,
    D3PL_heatflux_PFEM,
    D3PL_YPLUS_PFEM,
    D3PL_UINDEX PFEM,
    D3PL - VISCOUS PFEM,
    D3PL_RAND_r0 EM,
    D3PL_RAND_r10
    D3PL-RAND C10' EM,
    D3PL-ALPH\overline{A} PFEM,
    D3PL_CFL_PFEM,
    D3PL-CMU-PFEM,
    D3PL_TUR\overline{B}INTENS_PFEM
    };
/* vector variable names */
enum {
    D3PL_Velocity_INS=1000,
    D3PL_Vorticity_INS,
    D3PL_Velocity_\overline{CESE,}
    D3PL_Vorticity_CESE,
    D3PL_Momentum_\overline{CESE,}
    D3PL Momentum TS CESE,
    D3PL_E_field_radflow,
    D3PL_H_field_radflow,
    D3PL Current density FEM,
    D3PL_Electri\overline{C_field_\overline{FEM,}}\mathbf{\prime}=\mp@code{L}
    D3PL_Magnetic_field_FEM,
    D3PL_Lorentz_force_FEM,
    D3PL_Vector__potentīal_FEM,
    D3PL_Current_density_BEM,
    D3PL_Electric_field_BEM,
    D3PL-}Magnetic-field BEM
    D3PL_Lorentz_force_\overline{BEM,}
    D3PL_Vector_potential_BEM,
    D3PL-
    D3PL_Surface_magnetic_field,
    D3PL_Surface_Lorentz_force,
    D3PL_Velocity PFEM,
    D3PL_Vorticity_vect_PFEM,
    D3PL_particle_velocity,
    D3PL_Average_Velocity_PFEM,
    D3PL_H_field_BEM,
    D3PL_māgnetiz
    D3PL_NF_Velocity_P\overline{FEM,}
    D3PL_DR\overline{A}G_PFEM,
    D3PL_Shear_vect_PFEM
};
/* symmetric tensor variable names */
enum {
    D3PL_INS_VELOCITY_GRAD=2000
};
```

```
/* chemistry species variable names */
enum {
    D3PL_CHEM_SPECIES=3000
};
typedef struct _d3pnt {
    char * name;
    int id;
} D3PLOT_NAME_TABLE;
/* Identifiers for solver-mesh combinations */
enum {
    FEM_Q1Q0_INS_CFD=0,
    CES\overline{E}C=CD-NOD\overline{E},
    CESE_CFD_ELEMENT,
    CESE_CFD_ELEMENT_TS,
    RADF\overline{LOW_\overline{FULL,}}\mathbf{}/2
    RADFLOW_NODE,
    EM_FEMSTER_SOLID_INTEG_PTS,
    EM_FEMSTER_TSHEL\overline{L_INTE\overline{G}}P\textrm{PTS},
    EM_FEMSTER_SHELL_\overline{INTEG_\overline{P}TS,}
    EM_FEMSTER_SOLID_CENTRŌID,
    EM FEMSTER TSHEL\overline{L CENTROID,}
    EM_FEMSTER_SHELL_\overline{CENTROID,}
    EM_FEMSTER_AIR,
    RE\overline{CT}}\textrm{AIR EM}\mathrm{ NODE,
    EM_FEMST\overline{ER_D}\=\M,
    PFEM_IF,
    PFEM-IF SURFACE,
    STOC\overline{HASTIC PARTICLES,}
    CESE,
    CESE_SURFACE,
    EM,
    EM_SURFACE,
    EM_FEMSTER_SOLIDSHELL,
    EM_FEMSTER_NODE
};
static D3PLOT NAME TABLE d3plot_solver_name[] = {
    {"Incompressible-FEM CFD",FEM_Q1Q0_INTS_CFD},
    {"CESE CFD node",CESE_CFD_NODE\overline{}},
    {"CESE CFD element",CESS \overline{CFD ELEMENT},}
    {"CESE CFD taylor series",CE\overline{SE_CFD_ELEMENT_TS},}
```



```
    {"Radiation transport",RADFLOW_NODE},
    {"EM solid integ. pts",EM_FEMS\overline{TER_SOLID_INTEG_PTS},}
    {"EM tshell integ. pts",EM_FEMSTE\overline{R}_TSHEIL_INTEG_PTS},
    {"EM shell integ. pts",EM_\overline{FEMSTER_\overline{SHELL_INTEG_PTS},}}\mathbf{N}=,
    {"EM solid centroid",EM_F\overline{EMSTER_SO}LID_C\overline{ENTROI\overline{D}},}
    {"EM tshell centroid",EM_FEMSTE\overline{R}_TSHEİL_CENTROID},
    {"EM shell centroid",EM_\overline{FEMSTER__SHELL_CENNTROID},}
    {"EM solidShell centroi\overline{d}",EM_FEMSTER_\overline{SOLIDSHELL},}
    {"EM node",EM_FEMSTER_NODE},
    {"EM air",EM_\overline{FEMSTER_\overline{A}IR},}
    {"EM air - rēctangulār grid",RECT_AIR_EM_NODE},
    {"EM BEM",EM_FEMSTER_BEM},
    {"Incompressible CFD",PFEM_IF},
    {"Incomp. CFD surfaces",PFEM_IF_SURFACE},
    {"Stochastic particles",STOC\overline{H}AS\overline{TIC_PARTICLES},}
```

```
    {"CESE compressible CFD",CESE},
    {"Comp. CFD surfaces",CESE_SURFACE},
    {"EM nodes",EM},
    {"EM surface nodes",EM_SURFACE}
};
static D3PLOT NAME TABLE d3plot et name[] = {
    {"Pressure",
    {"Temperature",D\overline{3}PL_Temperature_INS},
    {"Enstrophy",D3PL Enstrophy INS},
    {"Helicity",D3PL \overline{Helicity INNS},}
    {"Stream function",D3PL_Stream_function_INS},
    {"Enthalpy",D3PL_Enthalpy_INS },
    {"Turbulent KE",就PL Turbūlent KE INS},
    {"Turbulent eps",D3P\overline{L}Turbulen\overline{t}}e\overline{p}s INS}
    {"Eddy Viscosity",D3PL_Eddy_Viscosity_INS},
    {"Density",D3PL Densit\overline{y INS}},
    {"Volume fractiō-1",D3\overline{P}L_VolFractSpec1_INS},
    {"Volume fraction-2",D3PL_VolFractSpec2_INS},
    {"Volume fraction-3",D3PL_VolFractSpec3-INS},
    {"Volume fraction-4",D3PL_VolFractSpec4_INS},
    {"Volume fraction-5",D3PL_VolFractSpec5_INS},
    {"Volume fraction-6",D3PL_VolFractSpec6_INS},
    {"Volume fraction-7",D3PL_VolFractSpec7-INS},
    {"Volume fraction-8",D3PL_VolFractSpec8_INS},
    {"Volume fraction-9",D3PL-VolFractSpec9-INS},
    {"Volume fraction-10",D3P\overline{L}}\mathrm{ VolFractSpec矛__INS},
    {"Fluid_velocity",D3PL_Velocity_INS},
    {"Vorticity",D3PL_Vorticity_INS},
    {"grad(velocity)",D3PL_INS_V\VLOCITY_GRAD}
    {"Density",D3PL_Density_CESEE},
    {"Pressure",D3P\overline{L_Pressure_CESE},}
    {"Temperature",D\overline{3}PL Tempe\overline{rature_CESE},}
    {"Total energy",D3P\overline{L}_Total_ener\overline{g}y_CESE},
    {"Enthalpy",D3PL_Enthalpy_\overline{CESE},}
    {"Entropy",D3PL_Entropy_CESE},
    {"Stream functiōn",D3PL_Stream_function_CESE},
    {"Void fraction",D3PL_void_fračtion_CESĒ},
    {"Schlieren_number",D\overline{3}PL_s\overline{chlieren_\overline{number_CESE},}}\mathbf{N}=
    {"Density Ta`ylor series",D3PL Densīty TS \overline{CESE},}
    {"Total energy Taylor series",D3PL_Total__energy_TS_CESE},
    {"Fluid_velocity",D3PL_Velocity_CESE},
    {"Vorticity",D3PL Vorticity CES\overline{E}},
    {"Momentum",D3PL_Momentum_CESE},
    {"Momentum Taylor series",D3PL_Momentum_TS_CESE},
    {"Temperature radflow",D3PL_Temperature_ra\overline{dflow},}
    {"Intensity radflow",D3PL_Intensity radflow},
    {"E-field radflow",D3PL_E_field_radflow},
    {"H-field radflow",D3PL H field radflow},
    {"Scalar potential",D3P\overline{L}_\overline{S}calar_potential},
    {"Electrical conductivity",D3PL_Electrical_conductivity},
    {"Ohm heating power FEM",D3PL_Ohm_heating_power_FEM},
    {"Ohm heating power BEM",D3PL_Ohm_heating_power_BEM},
    {"Current density FEM",D3PL_Cūrreñt_densi\overline{ty_FEM},}
    {"Electric field FEM",D3PL_Ēlectric_field_FEM},
    {"Magnetic field FEM",D3PL Magnetic field FEM},
    {"Lorentz force FEM",D3PL L̄orentz fōrce F\overline{EM}},
    {"Vector potential FEM",D\overline{3PL_Vector_potential_FEM},}
    {"Current density",D3PL_Currēnt_densisity_BEM},
    {"Electric field",D3PL_Electric_field_BEM},
    {"Magnetic field",D3PL_Magnetic_field_BEM},
    {"Lorentz force",D3PL_工_Lorentz_fōrce_B\overline{EM}},
    {"Vector potential",D\overline{3}PL_Vectōr_potēntial_BEM},
```

```
    {"Surface current",D3PL Surface current},
    {"Surface magnetic fiel\overline{d",D3PL_Surface_magnetic_field},}
    {"Surface Lorentz force",D3PL_Surface_Lorentz_force},
    {"Fluid velocity",D3PL_Velocity_PFEM},
    {"Fluid temperature",D\overline{3}PL_Temperrature_PFEM},
    {"Fluid pressure",D3PL_Pressure_PFEM},
    {"Fluid vorticity",D3P\overline{L_Vorticity_PFEM},}
    {"Fluid pressure",D3PL_\overline{Pressure_PFEM},}
    {"Turbulent K.E.",D3PL_K_PFEM},
    {"Turbulent eps.",D3PL_e\overline{ps_PFEM},}
    {"Particle size",D3PL \overline{partícle size},}
    {"Particle velocity",D3PL_particle_velocity},
    {"Particle temperature",D\overline{3PL_particle_temperature},}
    {"# of child particles",D3PL_particle_cnt_child_particles},
    {"Pressure Coefficient",D3PL_Cp_PFEM},
    {"Q Criterion",D3PL_Qc_PFEM},
    {"Surface Shear",D3\overline{PL_S_Shear_PFEM},}
    {"Level Set",D3PL_LEVELSET \overline{PFEM},}
    {"Viscosity",D3PL_VISCOUS_\overline{PFEM},}
    {"Average Pres",D\overline{3}PL AVG \overline{PRES PFEM},}
    {"Turbulent Visc",D3\overline{P}L_T\overline{U}RB_V\overline{I}SC_PFEM},
    {"H field",D3PL_H_field_BEM},
    {"Magnetization",\overline{D3PL_maggnetization_BEM},}
    {"Relative permeabilī
    {"Heat Trans. Coefficient",D3\overline{PL_HCC_PFEM},}
    {"Heat Flux",D3PL_heatflux_PFEM},
    {"Y plus",D3PL YP\overline{LUS PFEM},}
    {"Uniformity Iñdex",晾3PL_UINDEX_PFEM},
    {"Near Field Vel.", D3PL_NF_Velocity_PFEM},
    {"Drag Distribution", D3\overline{P}L_\overline{D}RAG_PFEM},
    {"Surface Shear Vector",D3\overline{P}L_Shēar_vect_PFEM},
    {"randle r0", D3PL_RAND_r0_EM}}
    {"randle r10", D3P\overline{L}_RAN\overline{D}_r\overline{10}_EM},
    {"randle c10", D3PL_RAND_c10_EM}
    {"CFL number",D3PL_CFL_PFEM},
    {"RANS Cmu",D3PL CMU P\overline{FEM},}
    {"Turbulent Inteñs",晾3PL_TURBINTENS_PFEM}
};
#endif
```


## DES CONTROL BLOCK AND DATA FORMAT

NPEFG - Word \#54 : when the seventh digit is set to 1 or 2 , there is a DES output. i.e. xx 1 xxxxxx or $\mathrm{V}=\mathrm{NPEFG} / 10000000=1$ or 2
$\mathrm{V}=1:$ version 1,8 words in the control block, with words 9 to $16=0$. (not used)
$\mathrm{V}=2$ : version 2,16 words in the control block.

## 1. EXTENDED MASTER CONTROL WORD

There will be one word as the master control word (NDEDB).
It defines the number of "Extended Control Blocks" in Section 2, and the same number of "State Data Blocks" in Section 3.

## 2. DES EXTENDED CONTROL BLOCK

2.1 First 8 words in the block are:

2.2 Output Variable Definitions:
$\mathrm{N}=\mathrm{NGPV}+\mathrm{NGPV}+\mathrm{NSPV}+\mathrm{NSEV}$
2.2.1 There will be N words of output type flags (integers) for each variable listed:

The data type flag is in a 6 -digit form: GGGCTF, where digits are as following:
$\mathrm{F}:=1$, for integers; $=2$, for real numbers; $=3$, for complex numbers
$=5$, for array of integers; $=6$, for array of reals; $=7$, array of complexes
$=8$, for combined data
$\mathrm{T}:=0$, for a scalar variable; $=1$, for a global vector; $=3$, for a global tensor $=5$, for a local vector; $=7$, for a local tensor; $=8$, for array data

Notes: A data block contains variable number and types of data, and is included only in the extended control block.
The first integer of a data block $(\mathrm{N})$ is the length of the data block, excluding the first integer. The definition of the data block identified by the group ID.
The post processor may ignore the data block by skipping these $(\mathrm{N}+1)$ words, if the definition of the data block is unknown.
C : The component ID of the grouped variables, defined by group ID, $=0$ : for a scalar variable.
$=1,2,3$ : components $(\mathrm{X}, \mathrm{Y}, \mathrm{Z})$ of a vector variable.
$=1,2,3,4,5,6$ : components (XX, YY, ZZ, XY, YZ, ZX) of a symmetric tensor
$=7,8,9$ : components ( $\mathrm{ZY}, \mathrm{XZ}, \mathrm{YX}$ ) of a spin tensor.
$=1,2,3,4,5,6,7,8,9$ : components (XX, YY, ZZ, XY, YZ, ZX, ZY, XZ, YX) of a general tensor.

Note: If any components of a vector or tensor not present it is zero.
If none of components $(7,8,9)$ are present, the tensor is symmetrical If only components $(7,8,9)$ are present, the tensor is unsymmetrical If components $(7,8,9)$ are present with other components the tensor is general.

GGG: The group ID of the grouped variables (GID)
2.2.2 There will be $8 * \mathrm{~N}$ words of the names output for variables listed after the type output (character*8).

### 2.3 DES GEOMETRY BLOCK

2.3.1 There will be NGPV blocks describing geometry data for each DES Part. The block size varies for each part, if variable data blocks are included, i.e. $1+$ Sum $\{\mathrm{NGPVi}+1, \mathrm{i}=2, \mathrm{NGPV}\}$.
word \#1: Part ID
word \#2: $\quad$ number of words reserved for a DE Part (NGPVi)
word \#3: $\quad$ NGPVi +2 : Reserved for a DE element
2.3.2 There will be NGEV blocks describing geometry data for DE Elements.

The block size varies for each element, if variable data blocks are included, i.e. $5+\mathrm{Sum}\{\mathrm{NGEVi}+1, \mathrm{i}=6, \mathrm{NGEV}\}$.
word \#1: Element ID
word \#2: Part ID
word \#3: Radius
word \#4: Mass
word \#5: Inertia
word \#6: number of words for reserved for a DE element (NGEVi)
word \#7: $\quad$ NGEVi+6: Reserved for a DE element

## 3. DES STATE DATA BLOCK

There are no arrays or data blocks in the state data.
3.1 Header for each state: current information for each part: there will be NDESP blocks of part geometry data for each part in.
Data output for each part in a state is determined by word \#7 (NSPV) from the DES control block. word \# 1: Number of active DE elements in each part.
to
word \# NSPV:
3.2 Following each state header: DE Element Data

Data for NDESE DE elements in this state will be listed. Each element will have NSEV (Word \#8) words of data output in the database.

The first variables should be always in the database
word \# 1: flag, 0 active, $<0$ inactive
to
word \# NSEV:

## D3ACS

The binary plot file D3ACS gives the results of frequency domain finite element acoustic analysis. The analysis is activated by the keyword:
*FREQUENCY_DOMAIN_ACOUSTIC_FEM.
The results of acoustic analysis are given as complex acoustic pressure at the nodes. The real and imaginary parts of the pressure, the magnitude of the pressure and a sound pressure level $(\mathrm{dB})$ are written to D3ACS as nodal state variables for each output frequency. No any element state variables are output.

## 1. Header file

1. File type $=26$
2. There are no element data, values: NV1D, NV2D, NV3D and NV3DT all $=0$
3. Nodal data flags are: $\mathrm{IU}=1, \mathrm{IT}=0, \mathrm{IV}=1$, and $\mathrm{IA}=1$.

## 2. State data

The state data is as follows for each frequency:
Time word = Frequency value
NGLBV as given for d3plot
NODEDATA $=(I T+N D I M *(I U+I V+I A)) * N U M N P$
CFDDATA $=0$
ELEMDATA $=0$
The ordering NODEDATA is as follows:
IU data: X, Y, Z Coordinates for each Node.
IV data: Real part, Imaginary part and Magnitude of Nodal Acoustic Pressure for each Node.
IA data: Sound Pressure Level (dB), Real part and Imaginary part of Nodal Normal Velocity for each Node.

